

# **AEROMODELLER POCKET DATA BOOK**

Originally published in America by FLYING MODELS this valuable collection of data sheets for aeromodellers provides almost a lifetime's collection of information on all aspects of model flying. Where necessary, facts have been amended to meet the requirements of British enthusiasts but many typical American expressions have been retained where international usage has accepted them.

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**MODEL & ALLIED PUBLICATIONS LTD**

**MAP**

**TECHNICAL PUBLICATION**

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M.A.P.

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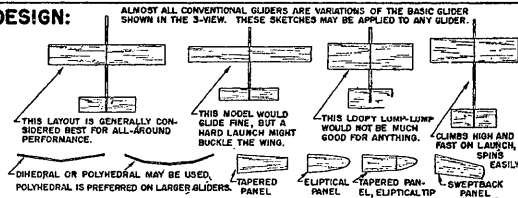
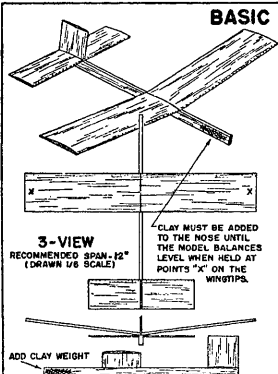
UNWIN BROTHERS LIMITED  
OLD WOKING, SURREY

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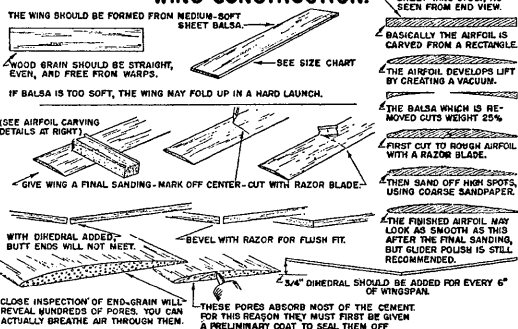
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# 1. BUILDING, FLYING & ADJUSTING ALL-BALSA GLIDERS

## BASIC DESIGN:



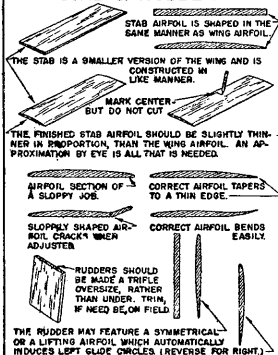
## WING CONSTRUCTION:



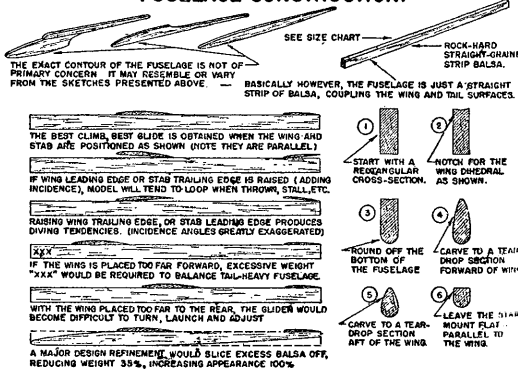
## MATERIAL CHART:

SPAN	CHORD	WING SH	LENGTH	FUSELAGE	TAIL SH
12"	2"	1/8"	10"	3/16" x 1/2"	1/8" SOFT
14"	2-1/4"	3/16"	11-3/4"	3/16" x 9/16"	1/8" SOFT
16"	2-1/2"	3/16"	13-1/2"	1/4" x 5/8"	3/32" SOFT
18"	2-3/4"	3/16"	15-1/4"	1/4" x 5/8"	1/8" SOFT
20"	3"	1/4"	17"	1/4" x 3/4"	1/8" SOFT

## STAB & RUDDER:



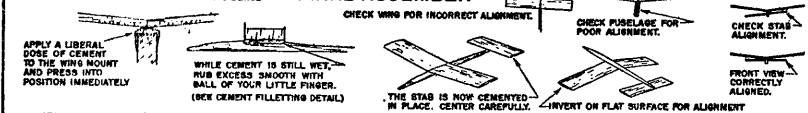
## FUSELAGE CONSTRUCTION:



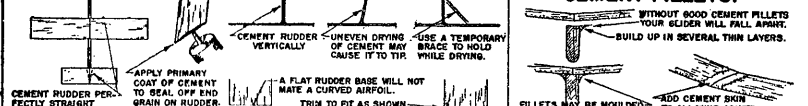
## FUSELAGE:



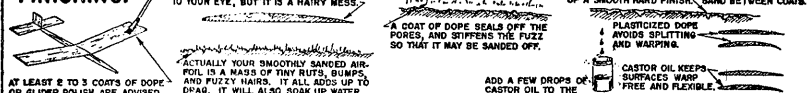
## FINAL ASSEMBLY:



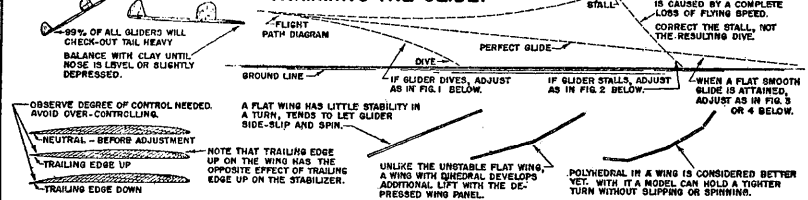
## CEMENT FILLETS:



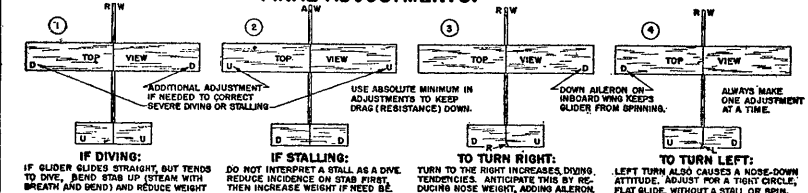
## FINISHING:



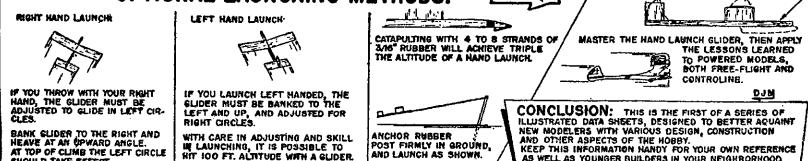
## TRIMMING THE GLIDE:



## FINAL ADJUSTMENTS:

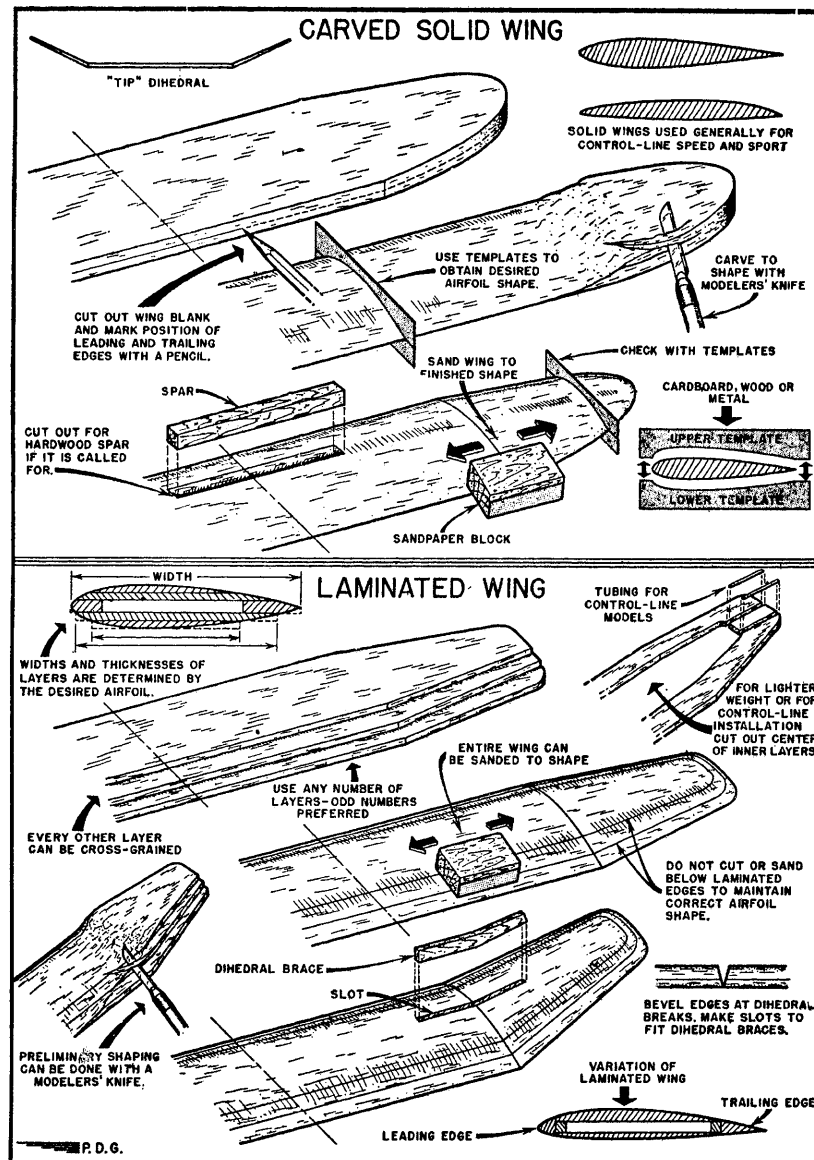
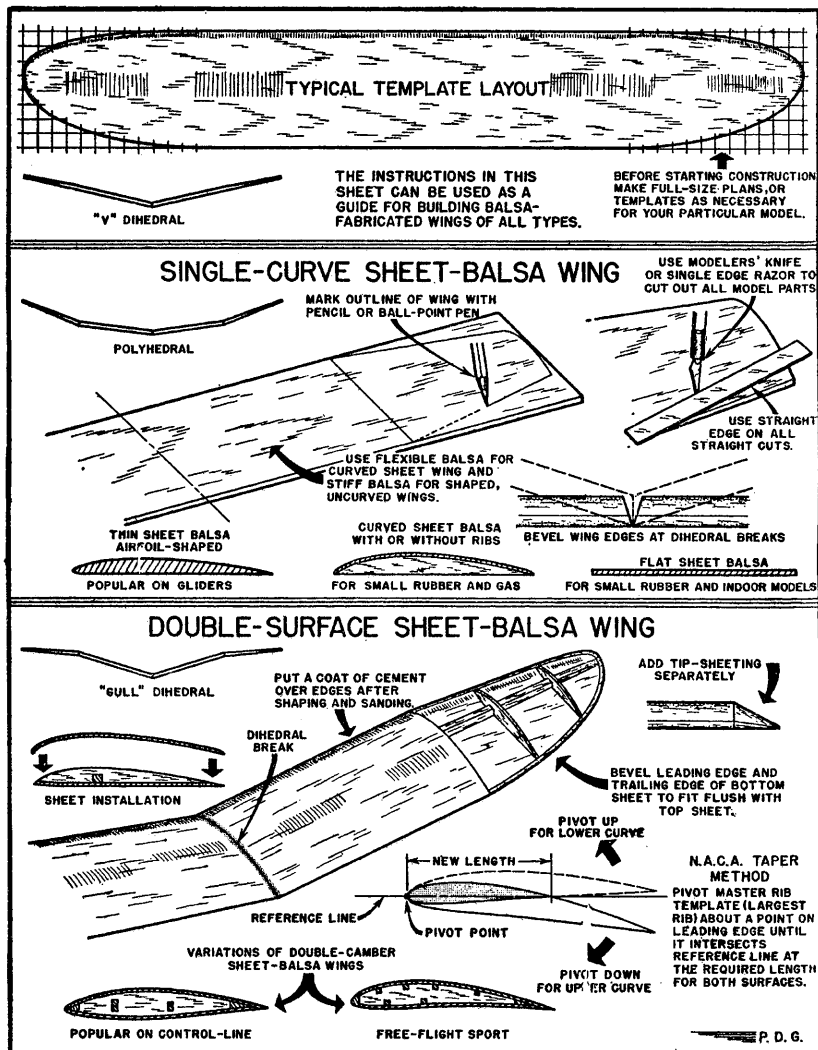


## OPTIONAL LAUNCHING METHODS:



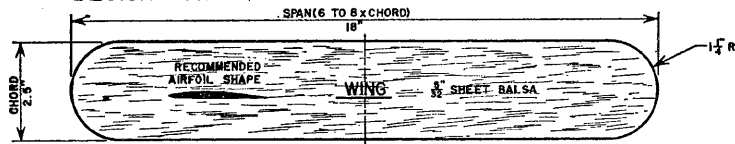


## 2. BALSA FABRICATED WINGS



### 3. CHUCK-GLIDER DESIGN DATA

#### DESIGN LAYOUT FOR BASIC HAND-LAUNCH GLIDER



**WING SIZE:** Select a span and chord which will give a wing size from 30 to 50 square inches in area. The span should be six to eight times the length of the chord; the approximate thickness of the wing airfoil,  $1/16"$  to each  $1"$  of chord.

**WING AREA:** To determine the amount of wing area, multiply the span by the chord; then subtract the area removed in making the semi-circular tips (Area of half-circle is determined by multiplying the radius by the radius and then by 1.57). An airfoil shape similar to the one shown above is recommended, as it will assist in achieving excellent results with maximum stability.

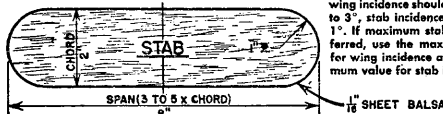


#### DIHEDRAL

$1\frac{1}{2}$  TO  $2^\circ$ /FT.

**FUSELAGE LENGTH:** The fuselage length—the total length of the model from the nose to the trailing edge of the stab or rudder, whichever is farther from the nose—should be approximately from three-quarters to the same length as the wingspan. The maximum fuselage depth should be no more than one-third the wing chord. The maximum fuselage thickness should not exceed one-quarter of the fuselage depth.

**MOMENT ARMS:** The distance between the centerline of the wing and the centerline of the stab is known as the Tail Moment Arm. This should be one and one-half to twice the length of the Nose Moment Arm (the distance between the centerline of the wing and the nose of the fuselage), with the length of both tail and nose moment arms equaling the fuselage length between the nose of the fuselage and the centerline of the stabilizer.



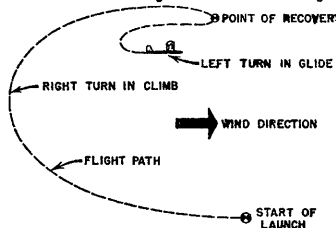
**STAB AREA:** The size of the stabilizer should range from 35% to 45% of the wing area. Also note that if the tail moment arm selected closely approximates the minimum suggested, then use a large stab area. If the tail moment arm selected is nearer the maximum suggested, then use a small stab area.

**ADJUSTING TECHNIQUE:** Add ballast in the form of clay to the nose of the fuselage until a glide is obtained with a slight nose-up tendency. For right-hand launch to the right adjust for a left turn. Begin by warping the rudder to the left a little at a time until a wide circle is obtained in the glide. Make final adjustments by warping right inboard wing panel down, or, if recovery is too quick and the model stalls, use right stab panel to delay recovery and tighten turn. In doing so, it may be necessary to remove some clay to compensate for the additional nose-down effect of the stab adjustments.

**INCIDENCE:** Incidence refers to the fixed angle at which the wing or stab is set with reference to the horizontal reference line of the fuselage side view (see drawing). For a basic hand-launch glider, wing incidence should be from  $2^\circ$  to  $3^\circ$ , stab incidence from  $0^\circ$  to  $1^\circ$ . If maximum stability is preferred, use the maximum value for wing incidence and the minimum value for stab incidence.

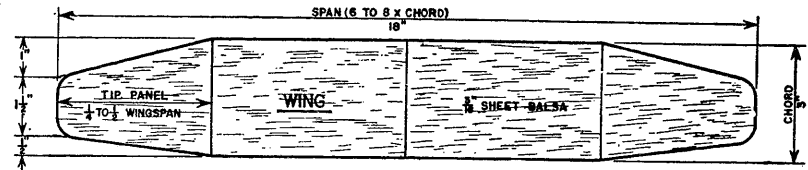
**RUDDER AREA:** Select the values which identify the type of rudder used. For a single rudder, the area should be 9% to 11% of the wing area; for a single rudder with tip plates, use 10% to 13%; for double rudders, 12% to 16%.

**CENTER OF GRAVITY:** The point at which the model airplane is balanced about all its axes is known as the Center of Gravity. When the center of gravity is too far forward, the model will dive, and when too far to the rear, the model will stall. If you follow the proportions set forth for the correct tail and nose moment arms for this glider, the theoretical center of gravity will be located approximately one-third of the wing chord forward of the trailing edge.



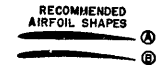
P.D.G.

#### DESIGN LAYOUT FOR CONTEST HAND-LAUNCH GLIDER

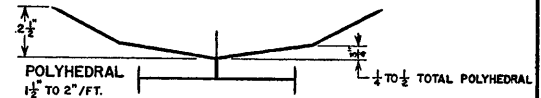


**WING SIZE:** Select a span and chord which will give a wing area of from 40 to 60 square inches. This size is preferable if your only previous experience has been with a basic hand-launch glider. Use the same chord-to-span ratio limits as on the basic hand-launch glider. A high span-to-chord ratio is preferable on a contest glider. The thickness ratio remains the same as on the basic hand-launch glider.

**WING AREA:** Determine the amount of wing area using the same procedure as on the basic hand-launch glider, subtracting the area cut away at the wing tips. For a tapered wing panel, figure the area of the cut-away triangles by multiplying one-half the length of each triangle by the width.



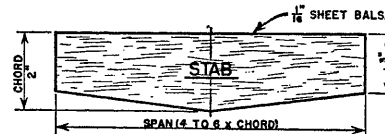
**AIRFOIL SHAPE:** There are two types of airfoil shapes which can be used. The flat bottom airfoil (A) is recommended as a first design after which an under-cambered section (B) can be employed if desired.



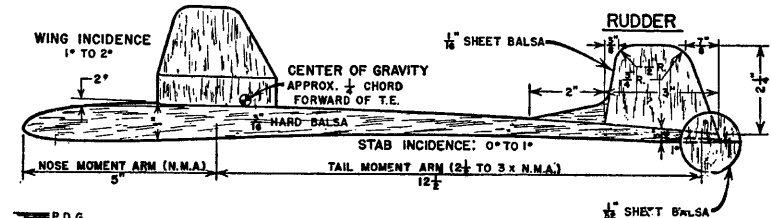
**INCIDENCE:** Only a small amount of incidence is recommended with a contest hand-launch glider to assist in maintaining a moderate amount of stability without affecting the altitude obtainable in launching.

**STAB AREA:** The stab should be 30% to 40% of the wing area. If previous experience with a contest hand-launch glider is lacking, or launching ability is moderate, use a stab area which closely approximates the maximum suggested, in conjunction with the minimum suggested for the tail moment arm.

**POLYHEDRAL:** Polyhedral, as shown, is a form of wing panel arrangement preferable to the V-type dihedral, except possibly when the span-to-chord ratio is low. In the design shown, polyhedral is used to best advantage for contest flying.



**FUSELAGE LENGTH AND DEPTH:** The fuselage length should be one to one and one-quarter the wing span. The maximum fuselage depth should be no more than one-third the wing chord. The maximum fuselage thickness should not exceed one-quarter of the fuselage depth.



P.D.G.

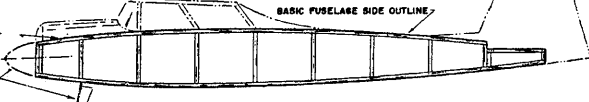
1/2 SHEET Balsa

## 4. SIMPLE SLAB-SIDER BUILDING

### STEP I—STUDY PLAN:

FAMILIARIZE YOURSELF WITH THE FUSELAGE PLAN, AND DETERMINE OUTLINE OF THE BASIC FUSELAGE SIDE TO BE CONSTRUCTED.

NOTE THE CABIN AND SUPERSTRUCTURE DETAILS ON THE SKETCH ARE ADDED LATER, AS ON MOST MODELS.



FUSELAGE SIDES SUCH AS THIS ONE, WITH ONE STRAIGHT AND ONE BENT LONGERON, HAVE A TENDENCY TO PULL OUT OF SHAPE. THIS CAN BE PREVENTED BY THE ADDITION OF DIAGONALS AS INDICATED.



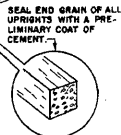
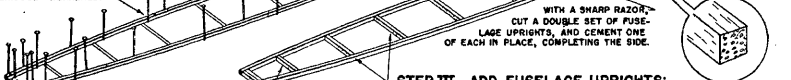
AN OPTIONAL METHOD OF INSTALLING DIAGONALS IS SHOWN HERE. IF 1/8" SQ. LONGERONS ARE USED, USE 1/16" SQ. DIAGONALS, NOTCHED INTO THE UPRIGHTS AS REQUIRED.

DIAGONALS ARE NOT REALLY NEEDED IF LONGERONS ARE BENT EQUALLY AS IN THESE SKETCHES.

### STEP II—LAY OUT LONGERONS:

SELECT MEDIUM-HARD STRAIGHT GRAINED WARP-FREE STRIPS OF Balsa FOR THE LONGERONS. BRACKET LONGERONS TO PLAN WITH PINS AS SHOWN.

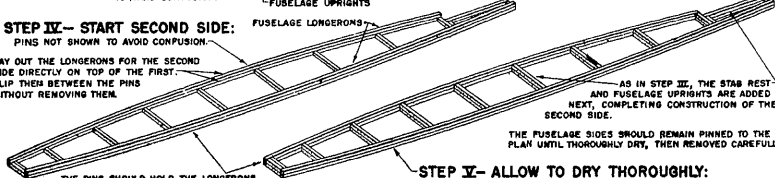
DO NOT PIERCE OR BRUISE LONGERONS WITH PINS, OR THEY WILL BE SERIOUSLY WEAKENED.



WITH A SHARP RAZOR, CUT A DOUBLE SET OF FUSELAGE UPRIGHTS, AND CEMENT ONE OF EACH IN PLACE, COMPLETING THE SIDE.

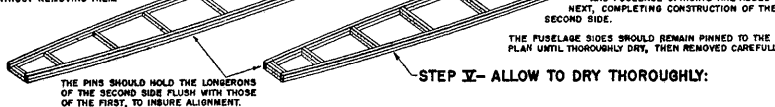
### STEP III—ADD FUSELAGE UPRIGHTS:

NOTE: PINS ARE NOT SHOWN TO AVOID CONFUSION.



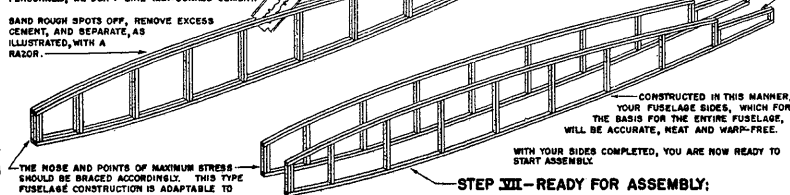
### STEP IV—START SECOND SIDE:

LAY OUT THE LONGERONS FOR THE SECOND SIDE DIRECTLY ON TOP OF THE FIRST. SLIP THEM BETWEEN THE PINS WITHOUT REMOVING THEM.



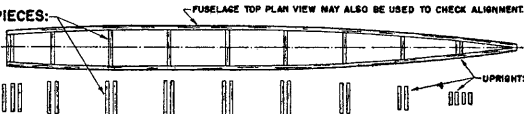
### STEP V—TRIM, SAND, AND SEPARATE:

WAX PAPER MIGHT PREVENT THE SIDES FROM STICKING SLIGHTLY, BUT THE CURE SEEMS WORSE THAN THE DISEASE. PERSONALLY, WE DON'T LIKE WAX-SOAKED CEMENT.



### STEP VI—CUT CROSS-PIECES:

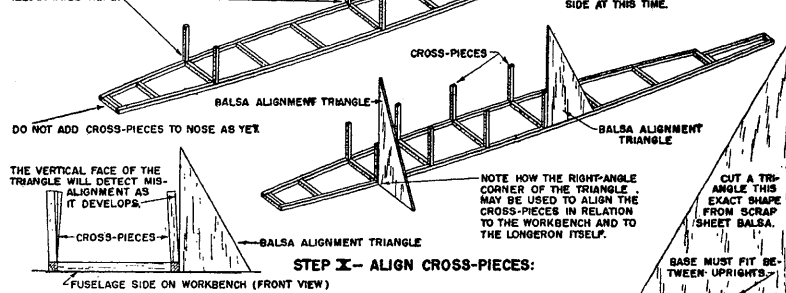
NEXT, FOCUS YOUR ATTENTION ON THE TOP PLAN VIEW. CAREFULLY CUT A SET OF CROSS-PIECES IN DUPLICATION AS INDICATED, AND PRECOAT END GRAIN WITH CEMENT AS ON THE FUSELAGE UPRIGHTS.



D.J.M.

### STEP II—CEMENT CROSS-PIECES:

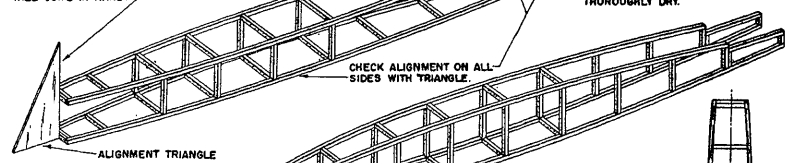
STARTING WITH THE STRAIGHTEST PORTION OF YOUR FUSELAGE, AS SEEN IN THE TOP PLAN VIEW, CEMENT THE REQUIRED UPRIGHTS TO THE FIRST SIDE AS ILLUSTRATED HERE.



### STEP I—ALIGN CROSS-PIECES:

### STEP II—CEMENT SECOND SIDE IN PLACE:

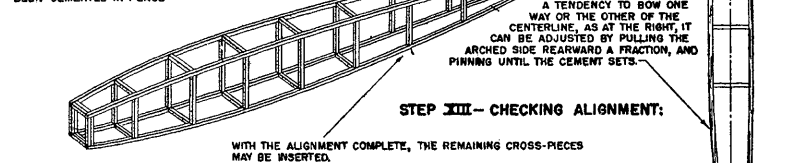
NEXT, APPLY A DROP OF CEMENT TO EACH CROSS-PIECE, AND DROP THE SECOND SIDE IN POSITION. ONCE AGAIN, THE Balsa ALIGNMENT TRIANGLE WILL COME IN HANDY.



### STEP III—NOSE CROSS-PIECES:

MOST FUSELAGES TAPER IN TOWARD THE NOSE, AS SEEN IN THE TOP VIEW AT THE EXTREME RIGHT.

TO SIMPLIFY THIS PART OF THE ASSEMBLY, STRETCH A RUBBER BAND AROUND THE NOSE, AFTER THE CROSS-PIECES HAVE BEEN CEMENTED IN PLACE.

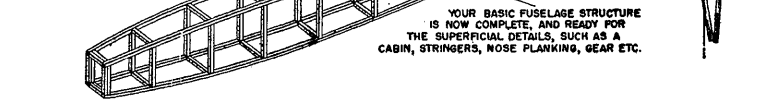


### STEP III—CHECKING ALIGNMENT:

WITH THE ALIGNMENT COMPLETE, THE REMAINING CROSS-PIECES MAY BE INSERTED.

### STEP IV—SANDING:

GIVE THE ENTIRE STRUCTURE A GENERAL SANDING TO REMOVE ROUGH EDGES, FINGERPRINTS AND EXCESS CEMENT.

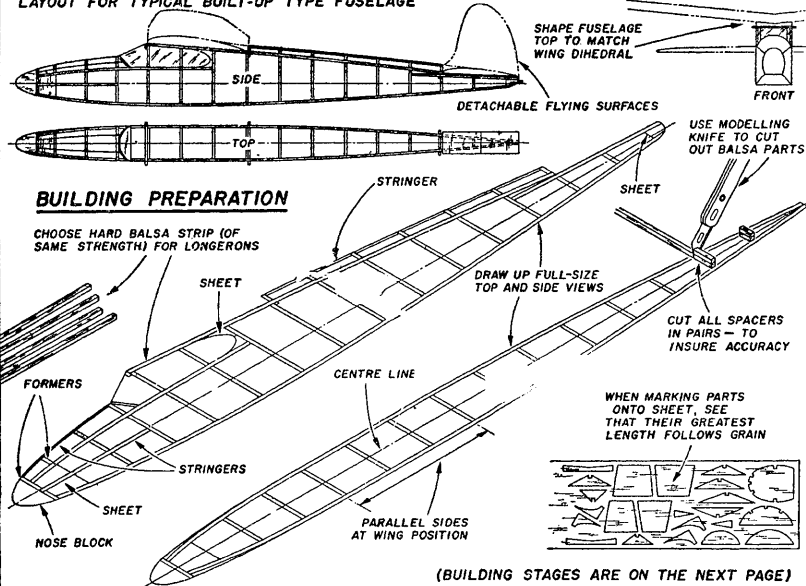


CONSTRUCTION PRINCIPLES OUTLINED HERE ARE APPLICABLE TO MOST MODELS USING THIS TYPE FUSELAGE CONSTRUCTION.

D.J.M.

## 5. CABIN SLAB-SIDERS

### LAYOUT FOR TYPICAL BUILT-UP TYPE FUSELAGE

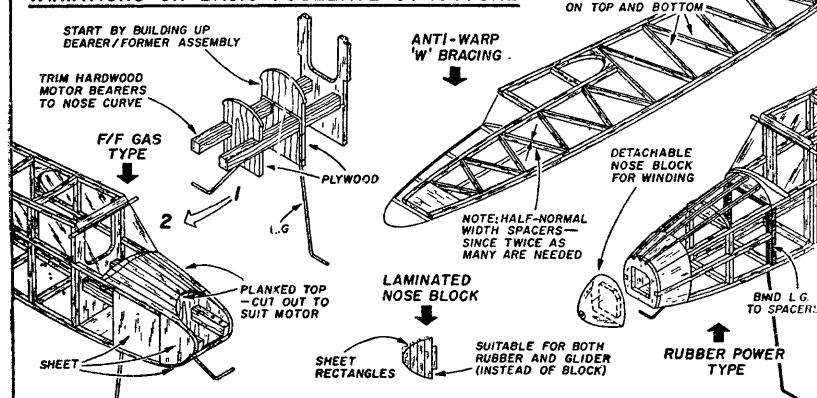


### BUILDING PREPARATION

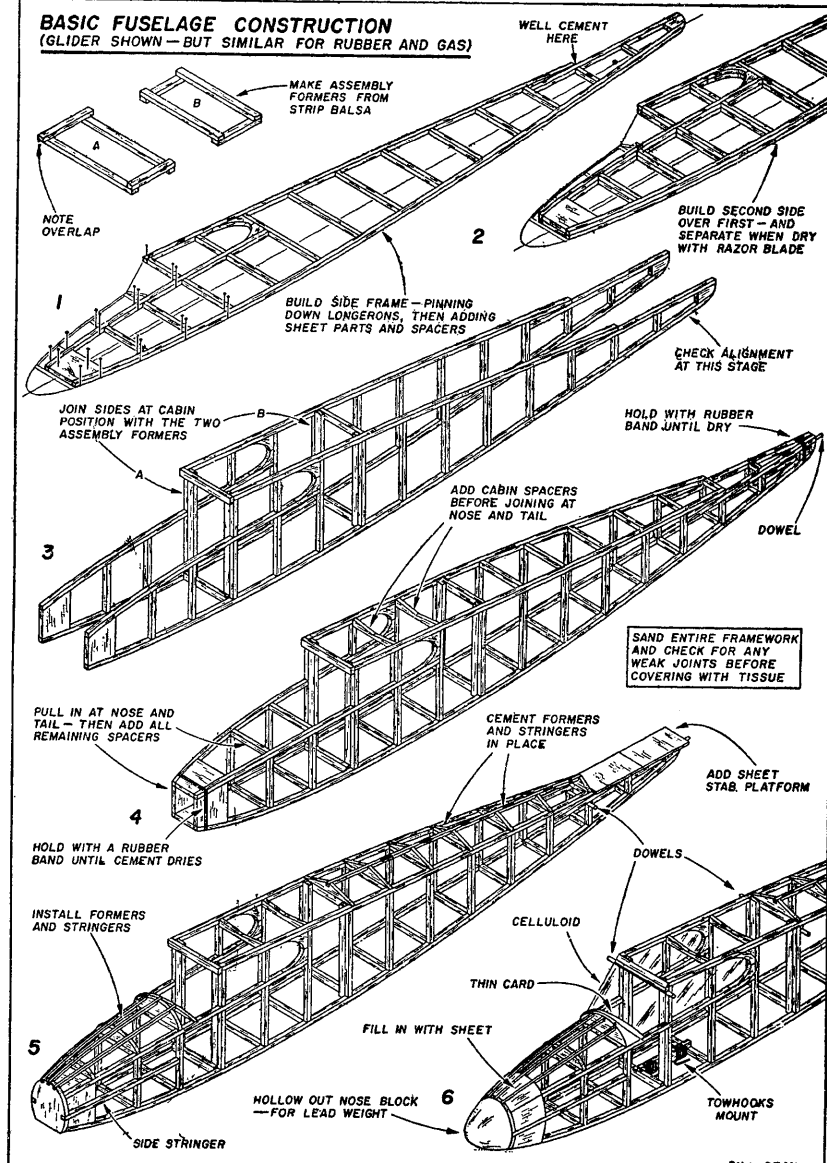
CHOOSE HARD Balsa STRIP (OF SAME STRENGTH) FOR LONGERONS

(BUILDING STAGES ARE ON THE NEXT PAGE)

### VARIATIONS ON BASIC FUSELAGE STRUCTURE



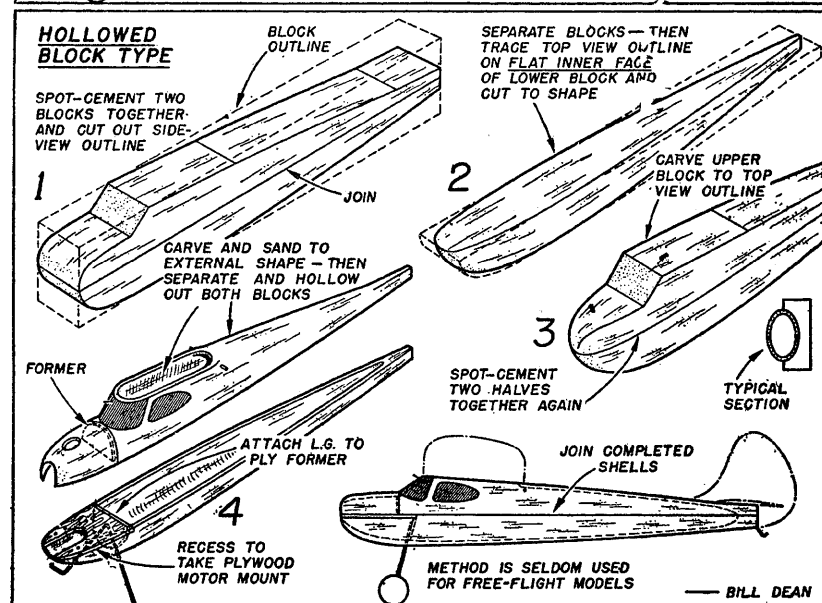
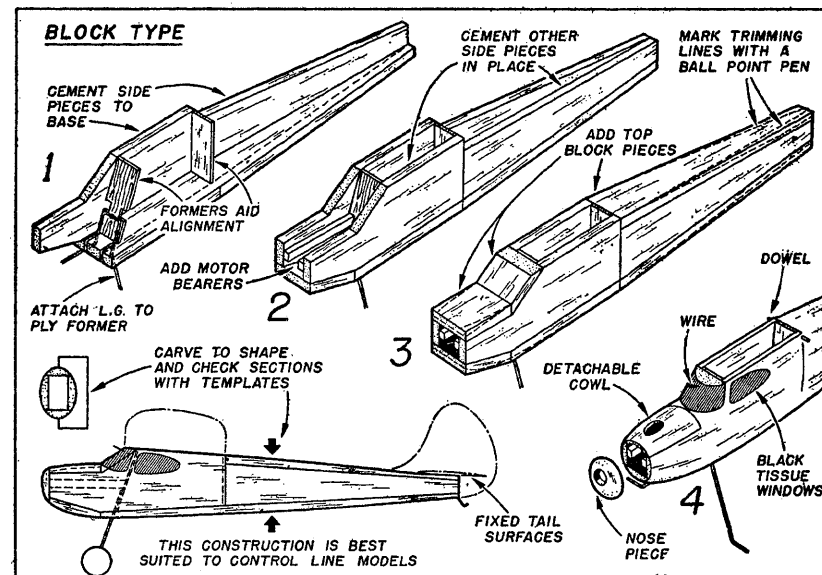
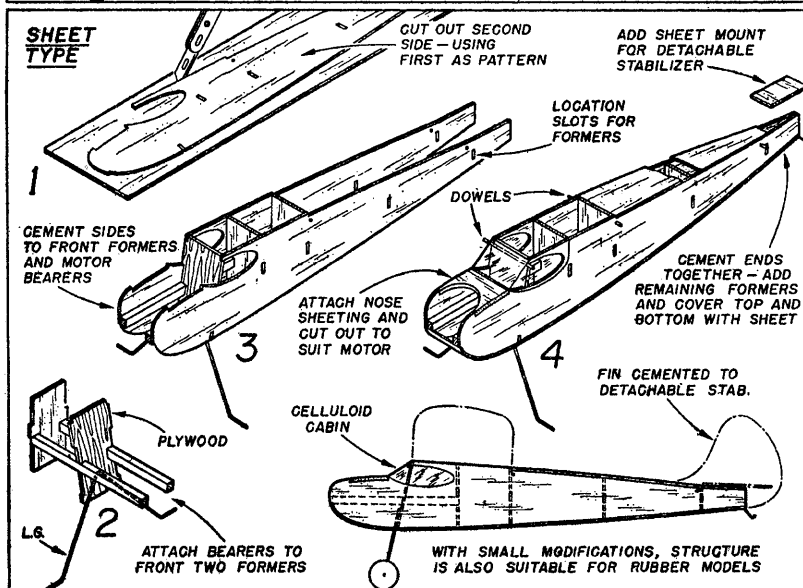
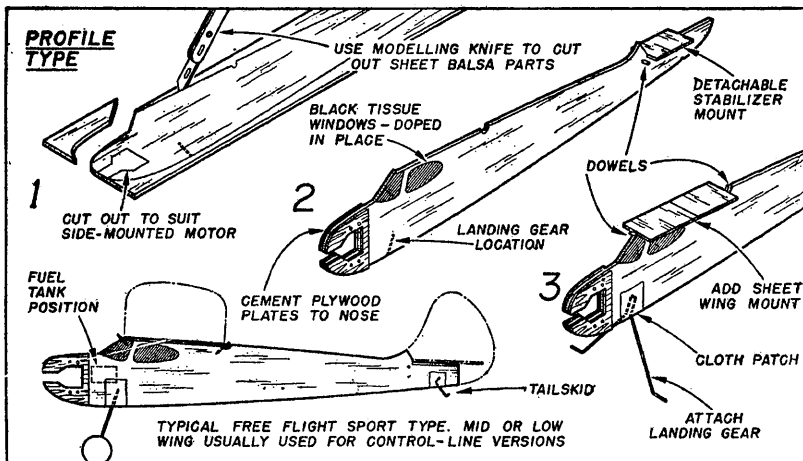
### BASIC FUSELAGE CONSTRUCTION (GLIDER SHOWN - BUT SIMILAR FOR RUBBER AND GAS)



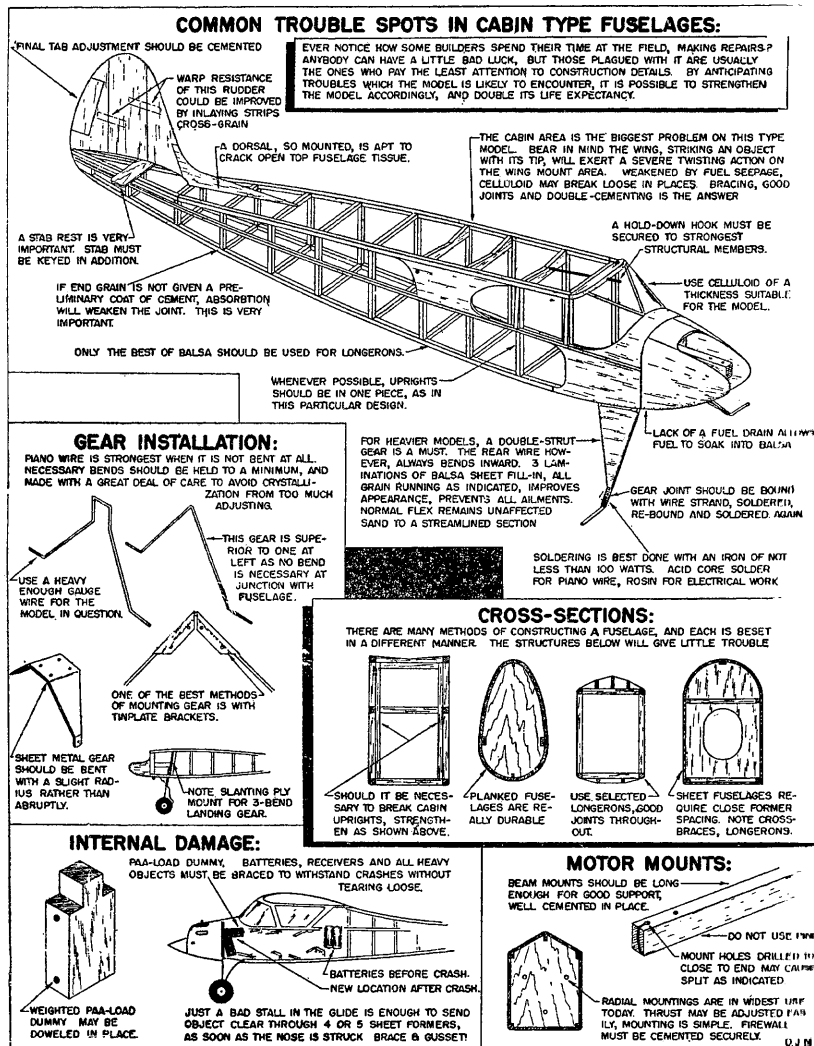
—BILL DEAN.



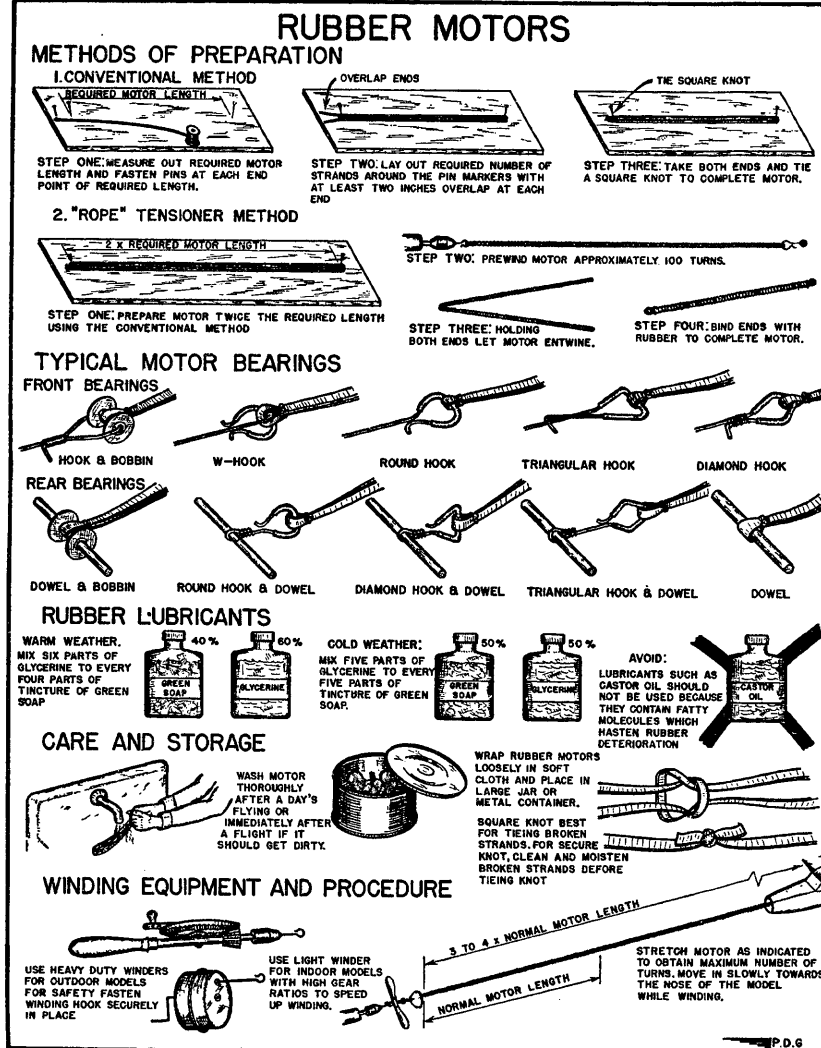
## 6. SHEET & BLOCK FUSELAGES



## 7. MAKING STRONGER FUSELAGES



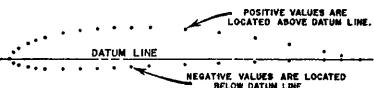
## 8. RUBBER MOTOR TECHNIQUE



## 9. AIRFOIL PLOTTING

### AIRFOIL TERMS-WHAT THEY MEAN:

UPPER CAMBER: THE UPPER CURVATURE OF THE AIRFOIL. LOWER CAMBER: THE LOWER CURVATURE OF THE AIRFOIL. UNDER CAMBER: THE REVERSE CURVATURE OF THE AIRFOIL.



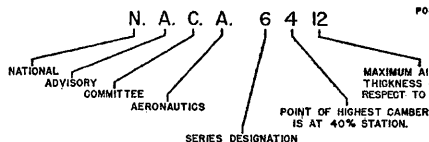
DATUM LINE: A REFERENCE LINE ABOVE AND BELOW WHICH THE POINTS FOR THE AIRFOIL CURVE ARE PLOTTED.

STATIONS: POSITIONS AT WHICH THE POINTS FOR THE AIRFOIL CURVE ARE TO BE PLOTTED. VALUES ARE ACTUALLY PERCENTAGES OF THE AIRFOIL LENGTH MEASURED FROM THE LEADING EDGE OF THE AIRFOIL.

N.A.C.A. 6412																		
STATION	0	1.25	2.5	5.0	7.5	10	15	20	25	30	40	50	60	70	80	90	95	100
UPPER CAMBER	0.00	2.73	3.80	5.38	6.57	7.58	8.18	10.34	11.14	11.65	11.80	11.18	9.95	8.23	6.03	3.33	1.79	0.12
LOWER CAMBER	0.00	-1.23	-1.84	-1.99	-2.05	-1.99	-1.87	-1.25	-0.76	-0.36	0.20	0.59	0.70	0.85	0.73	0.39	0.16	-0.12
LEADING EDGE RADIUS: 1.58		SLOPE OF RADIUS: 6/20																

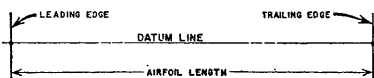
AIRFOIL ORDINATES: EACH AIRFOIL HAS A TABLE OF VALUES WHICH WHEN TRANSFERRED TO THE CORRESPONDING STATIONS ON THE LAYOUT, ABOVE AND BELOW THE DATUM LINE, WILL INDICATE THE POINTS FOR THE AIRFOIL CURVE. THE TOTAL AIRFOIL LENGTH IS CONSIDERED AS 100% WITH THE TABLE OF VALUES DIRECTLY RELATED. EXAMPLE: -1.23 = 1.23% LOCATED BELOW DATUM LINE AT THE SPECIFIED STATION.

### N.A.C.A. CODE BREAKDOWN

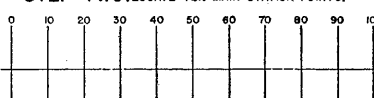


### PLOTTING THE AIRFOIL

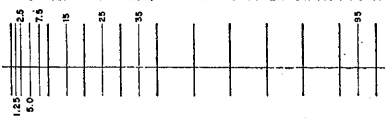
STEP ONE: SELECT AN AIRFOIL LENGTH, TRYING TO USE A SIZE WHICH CAN BE EASILY SUBDIVIDED INTO TEN MAIN STATIONS AND WHERE REQUIRED INTO SUBSTATIONS. EXAMPLE: A 6" AIRFOIL LENGTH WILL BREAK DOWN INTO TEN MAIN STATIONS  $\frac{6}{10}$  APART, ONE PERCENT EQUALLING  $\frac{1}{100}$ .



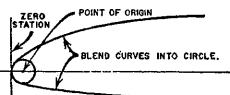
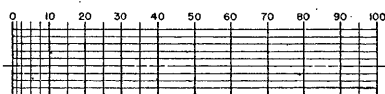
STEP TWO: LOCATE TEN MAIN STATION POINTS.



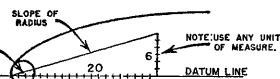
STEP THREE: LOCATE ALL OTHER STATION POINTS.



STEP FOUR: DRAW GRID LINES PARALLEL TO, ABOVE AND BELOW DATUM LINE AND SPACED APART 1% - 2% OF THE AIRFOIL LENGTH. NOTE: THIS STEP OPTIONAL, BUT ADVISABLE FOR MODELERS WITHOUT PREVIOUS PLOTTING EXPERIENCE.

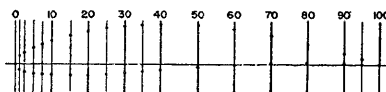


LEADING EDGE RADIUS: GENERALLY THE DISTANCE ON THE DATUM LINE FROM THE POINT OF ORIGIN OF THE CIRCLE TO THE ZERO STATION (LEADING EDGE). ONLY EXCEPTION BEING WHEN A SLOPE OF RADIUS IS GIVEN. NOTE: NOT ALL AIRFOILS HAVE A LEADING EDGE RADIUS.

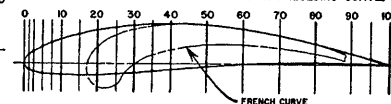


SLOPE OF RADIUS: WHEN A SLOPE IS GIVEN FOR THE LEADING EDGE RADIUS, IT IS TO INDICATE THAT ITS CENTER IS NOT LOCATED ON THE DATUM LINE BUT RATHER ON A DIAGONAL LINE REFERRED TO AS THE SLOPE OF THE RADIUS. A PROPORTION IS GIVEN AS IN THE N.A.C.A. 6412 FOR ITS CONSTRUCTION.

STEP FIVE: USING TABLE OF ORDINATES LOCATE VALUES FOR UPPER AND LOWER CAMBER AT DESIGNATED STATIONS. NOTE: HORIZONTAL GRID LINES OMITTED FOR CLARITY.



STEP SIX: USING FRENCH CURVES CONNECT THE PLOTTED POINTS TO OBTAIN THE AIRFOIL OUTLINE. SELECT CURVES THAT WILL CONNECT AT LEAST FOUR STATIONS AND BLEND SMOOTHLY WITH PRECEDING CURVE.

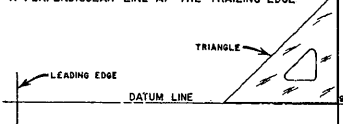


### PLOTTING THE AIRFOIL (SECONDARY METHOD FOR NON-DIVISIBLE LENGTHS)

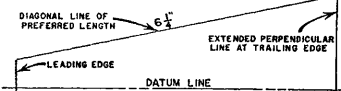
STEP ONE: SELECT DESIRED AIRFOIL LENGTH AND DRAW DATUM LINE



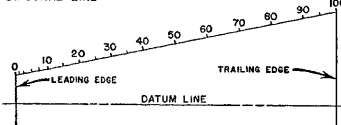
STEP TWO: WITH THE AID OF A TRIANGLE ERECT A PERPENDICULAR LINE AT THE TRAILING EDGE



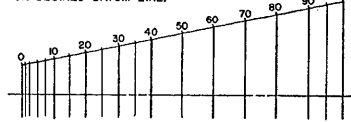
STEP THREE: DRAW LINE FROM POINT ABOVE DATUM LINE AT THE LEADING EDGE TO THE PERPENDICULAR LINE ERECTED AT THE TRAILING EDGE, USING A LENGTH WHICH CAN BE EVENLY SUB-DIVIDED.



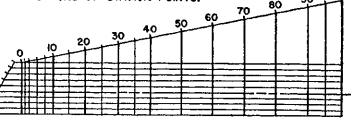
STEP FOUR: PLOT STATION POINTS ON THE DIAGONAL LINE



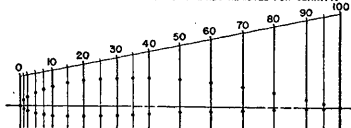
STEP FIVE: WITH THE AID OF A TRIANGLE DROP PERPENDICULAR LINES FROM STATION POINTS ON DIAGONAL LINE TO OBTAIN CORRECT POSITIONS OF STATION POINTS ON DESIRED DATUM LINE.



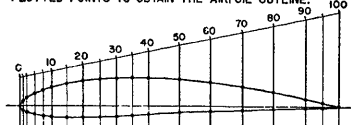
STEP SIX: SPACE GRID LINES 1% - 2% APART USING PROCEDURE OUTLINED FOR OBTAINING CORRECT POSITIONS OF STATION POINTS.



STEP SEVEN: USING TABLE OF ORDINATES LOCATE VALUES FOR UPPER AND LOWER CAMBER AT DESIGNATED STATIONS. NOTE: HORIZONTAL GRID LINES REMOVED FOR CLARITY.



STEP EIGHT: USING FRENCH CURVES CONNECT THE PLOTTED POINTS TO OBTAIN THE AIRFOIL OUTLINE.



### SIX POPULAR AIRFOILS

CLARK Y  
RECOMMENDED FOR FREE FLIGHT GAS, RUBBER, TOWLINE AND U-CONTROL SPORT OR BASIC DESIGNS

R.A.F 32  
RECOMMENDED FOR CONTEST FREE FLIGHT GAS, RUBBER AND TOWLINE DESIGNS

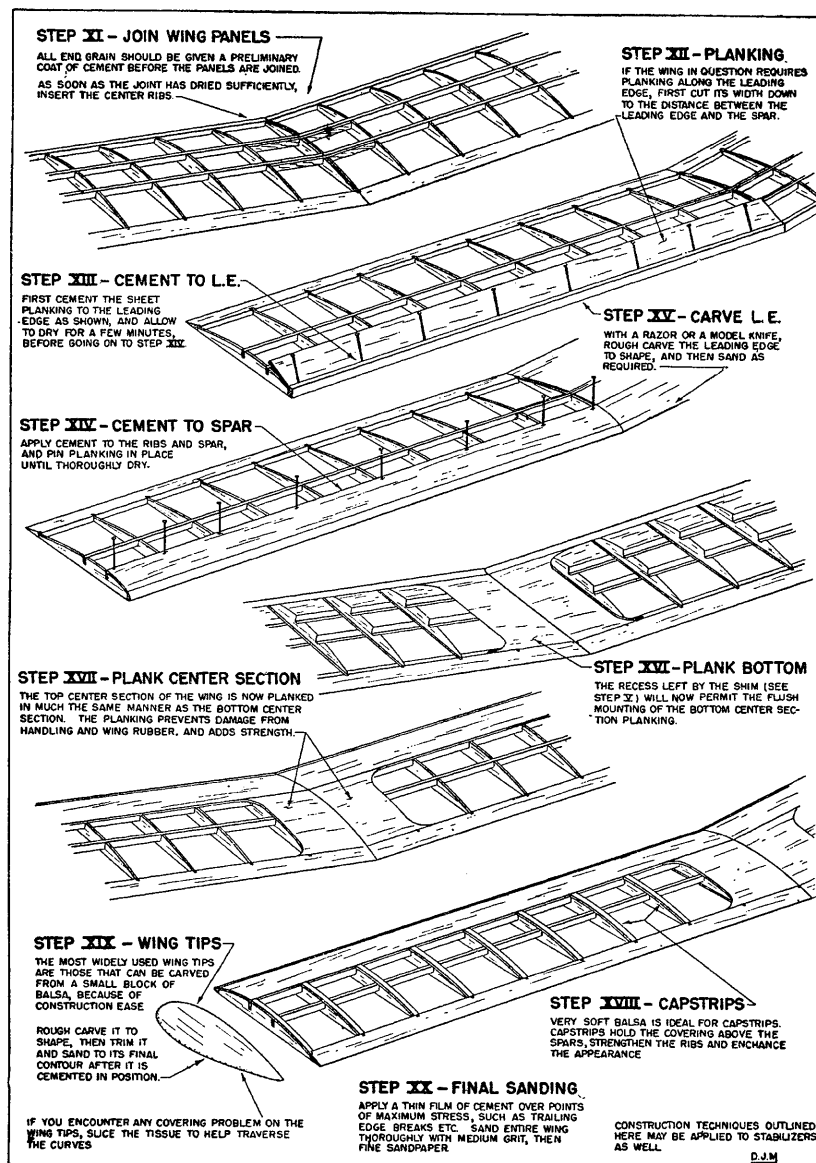
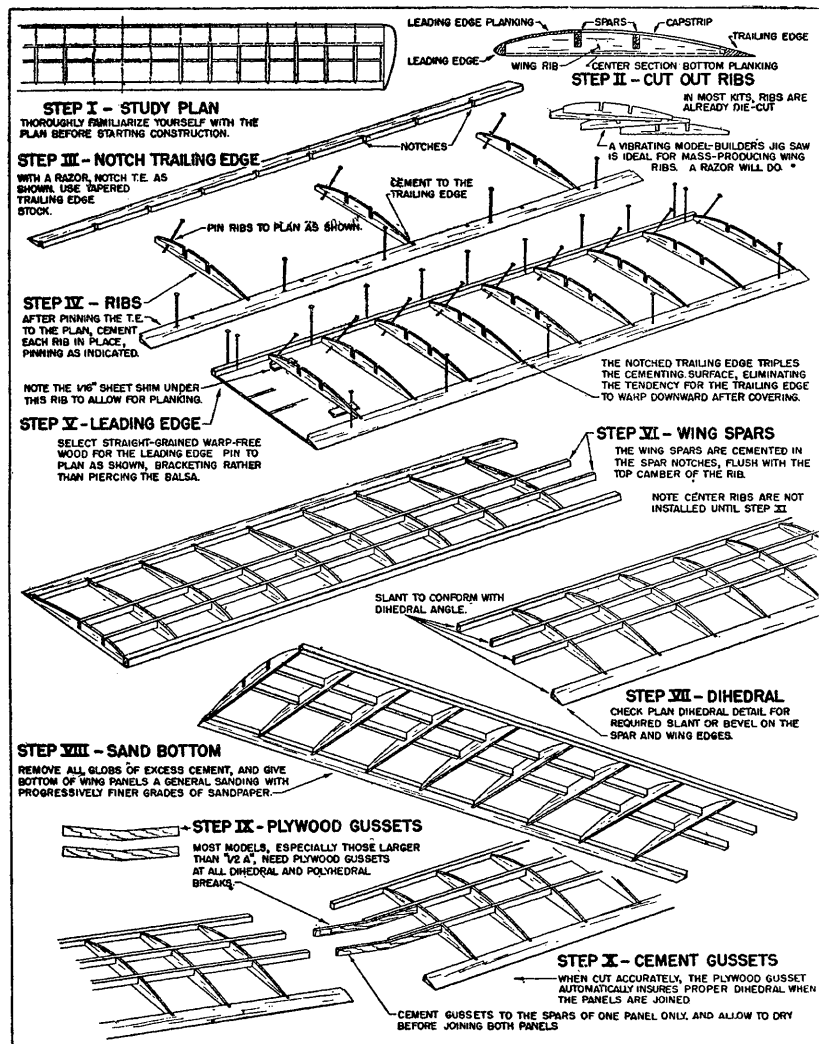
N.A.C.A. 6409  
RECOMMENDED FOR CONTEST FREE FLIGHT GAS, RUBBER AND TOWLINE DESIGNS

GRANT X-8  
RECOMMENDED FOR CONTEST FREE FLIGHT GAS, RUBBER, TOWLINE AND RADIO CONTROL DESIGNS

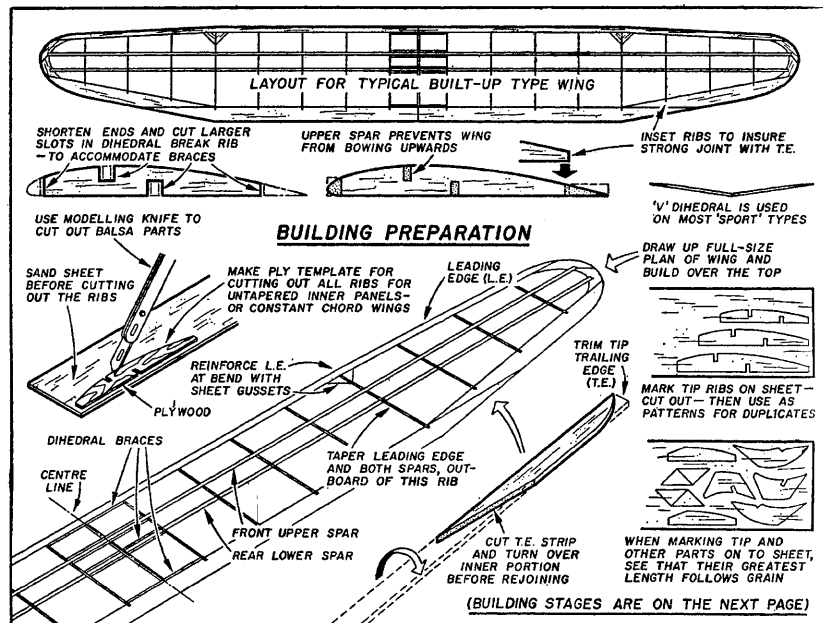
DAVIS  
RECOMMENDED FOR SPORT, BASIC AND CONTEST FREE FLIGHT GAS, RUBBER, TOWLINE AND FOR U-CONTROL SPORT AND SPEED DESIGNS

N.A.C.A. 2409  
RECOMMENDED FOR U-CONTROL SPORT AND SPEED DESIGNS

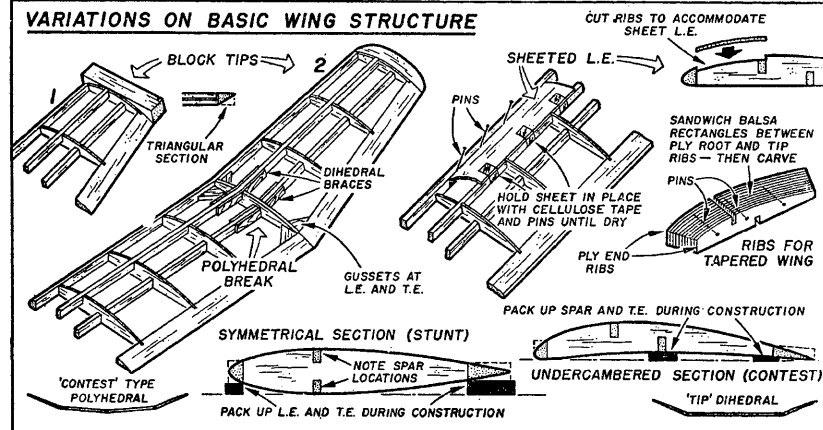
# 10. BASIC WING CONSTRUCTION



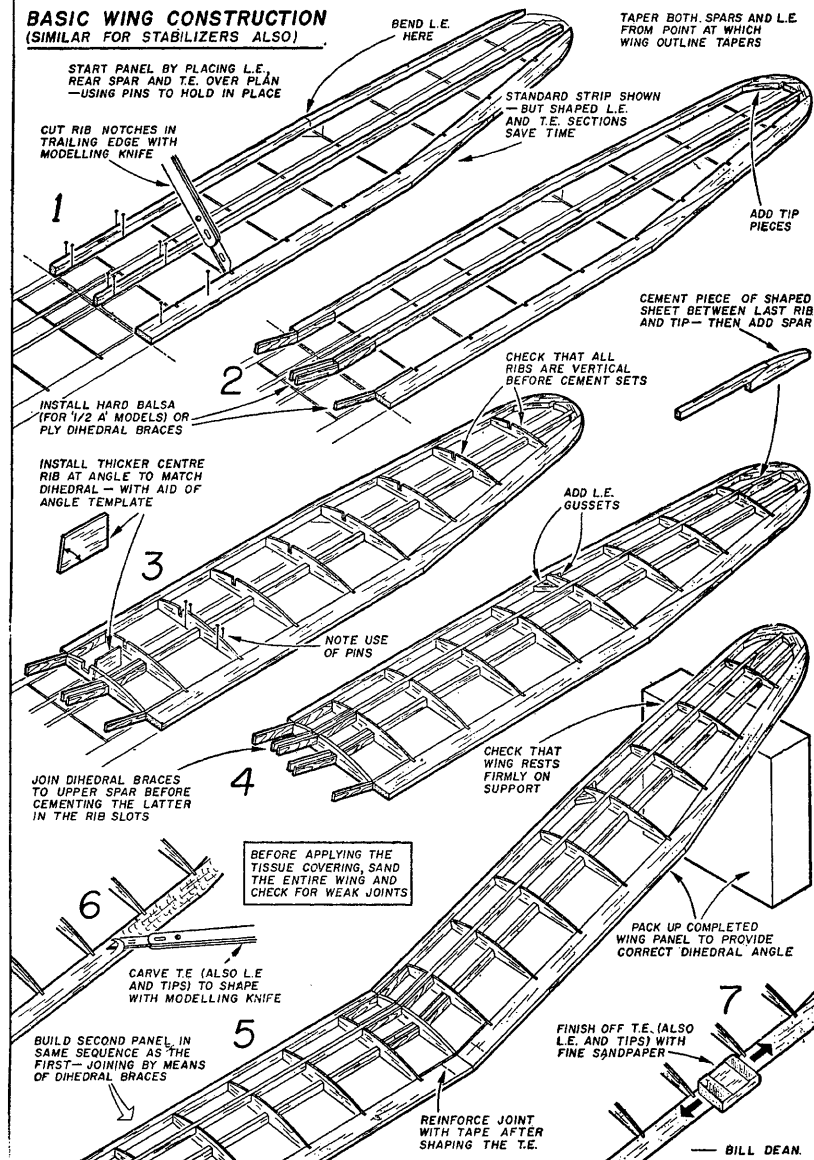
# 11. MORE COMPLEX WING BUILDING



## VARIATIONS ON BASIC WING STRUCTURE



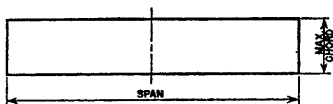
## BASIC WING CONSTRUCTION (SIMILAR FOR STABILIZERS ALSO)



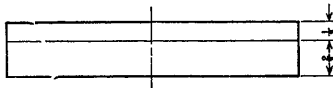


## CONSTRUCTION OF AN ELLIPTICAL PLANFORM

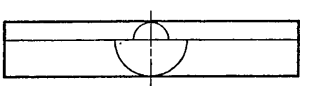
**STEP ONE:** CONSTRUCT RECTANGLE WHICH WILL ENCLOSE PROPOSED ELLIPTICAL PLANFORM. (MAXIMUM CHORD AND SPAN)



**STEP TWO:** DIVIDE RECTANGLE INTO TWO SECTIONS—ONE THIRD CHORD FOR LEADING EDGE SECTION AND TWO THIRDS CHORD FOR TRAILING EDGE SECTION. (THIS ARRANGEMENT WILL PRODUCE THE MOST POPULAR FORM OF ELLIPTICAL PLANFORM BEING USED. HOWEVER, THE SECTIONS CAN BE DIVIDED EQUALLY, REVERSED OR ALTERED IN ANY MANNER TO PRODUCE A GREAT MANY OTHER VARIATIONS OF THE ELLIPTICAL PLANFORM.)

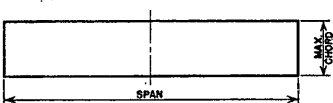


**STEP THREE:** DRAW TWO HALF CIRCLE ARCS FROM POINT OF INTERSECTION OF CENTERLINE, TANGENT (TOUCHING) TO THE LEADING AND TRAILING EDGES RESPECTIVELY.

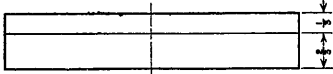


## CONSTRUCTION OF A PARABOLIC PLANFORM

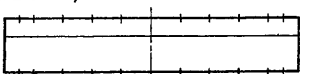
**STEP ONE:** CONSTRUCT RECTANGLE WHICH WILL ENCLOSE PROPOSED PARABOLIC PLANFORM. (MAXIMUM CHORD AND SPAN)



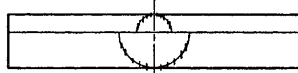
**STEP TWO:** DIVIDE RECTANGLE INTO TWO SECTIONS—ONE THIRD CHORD FOR LEADING EDGE SECTION AND TWO THIRDS CHORD FOR TRAILING EDGE SECTION. (THIS ARRANGEMENT AS IN ELLIPTICAL PLANFORM—CAN ALSO BE ALTERED TO PRODUCE OTHER VARIATIONS.)



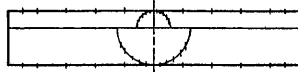
**STEP THREE:** DIVIDE SPAN OF RECTANGLE INTO EQUAL UNITS, SUBDIVIDING LAST REMAINING UNIT.



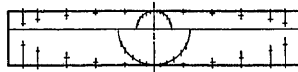
**STEP FOUR:** DIVIDE HALF CIRCLE ARCS INTO EQUAL UNITS, SUBDIVIDING LAST REMAINING UNIT. FOR SMALL PLANFORMS FOUR TO FIVE UNITS ARE SATISFACTORY, WHILE FOR LARGER PLANFORMS AS MUCH AS EIGHT TO TEN UNITS ARE RECOMMENDED FOR OBTAINING AN ACCURATE ELLIPTICAL PLANFORM.



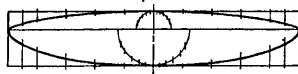
**STEP FIVE:** DIVIDE THE SPAN OF RECTANGLE INTO THE SAME NUMBER OF UNITS AS THAT OF EACH HALF CIRCLE ARC.



**STEP SIX:** CONNECT CORRESPONDING UNIT STATIONS BY PROJECTING THEM UNTIL THEY INTERSECT, PRODUCING POINTS ON THE ELLIPTICAL PLANFORM.



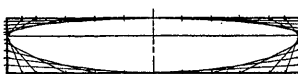
**STEP SEVEN:** USING FRENCH CURVES CONNECT PLOTTED POINTS TO OBTAIN ELLIPTICAL PLANFORM.



**STEP FOUR:** DIVIDE EACH SECTION OF THE CHORD INTO HALF THE NUMBER OF UNITS ON THE SPAN OF RECTANGLE.



**STEP FIVE:** CONNECT CORRESPONDING UNIT STATIONS BY PROJECTING DIAGONAL LINES.



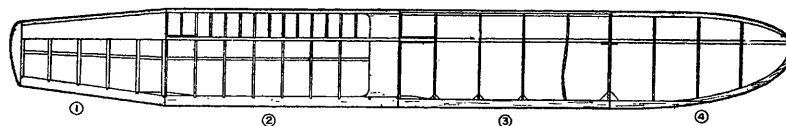
**STEP SIX:** USING FRENCH CURVES DRAW IN PARABOLIC PLANFORM BY FOLLOWING THE INSIDE PATTERN OBTAINED FROM DRAWING THE DIAGONAL LINES.



P.D.G.

## WING STRUCTURAL PROBLEMS:

THE GENERAL TENDENCY TODAY IS TO CONSTRUCT A WING FROM AS FEW PIECES AS POSSIBLE. EXPERIENCED WAKEFIELD BUILDERS AND SUCH CUT CORNERS TO SAVE WEIGHT, BUT THEY STILL RETAIN THEIR DESIRED AIRFOIL, RIGIDITY, AND NECESSARY STRENGTH-WEIGHT RATIO. THE OTHER 98% PREFER A SCANTY STRUCTURE AS IT IS THE PATH OF LEAST RESISTANCE, AND THAT IS WHEN THE TROUBLE STARTS.



IF UPON GLANCING AT PANELS 1 AND 2, YOU FEEL THE STRUCTURE IS TOO MUCH BOTHER, THEN YOU ARE ONE OF THE 98%. PANELS 3 AND 4 ARE THE ONES THAT ARE THE REAL BOTHER, FOR THE WARP RESISTANCE, STRENGTH AND AERODYNAMIC QUALITIES WILL BE POOR. EVEN TWO THE INITIAL BUILDING TIME OF PANELS 1 OR 2 MAY BE A FEW MINUTES MORE. THEY WILL STAND UP AGAINST THE RIGORS OF ACTIVE FLYING. CONSTRUCTION DEFECTS AND ADVANTAGES OF EACH PANEL ARE ITEMIZED BELOW.

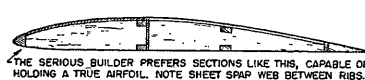
PANEL 1	PANEL 2	PANEL 3	PANEL 4
SHEETED LEADING EDGE MAINTAINS SMOOTH AIRFOIL. CAP STRIPS ON RIBS PREVENT BOWING AS VISIBLE IN PANEL 2. NOTE GUSSET AT TRAILING EDGE. EXCELLENT.	VERY GOOD SUBSTITUTE FOR CONSTRUCTION IN PANEL 1. EASIER TO REPAIR. TOP SPAR PREVENTS PANEL FROM ARCHING UPWARD. NOTCHED T.E. EXCELLENT.	POORLY SUPPORTED THIN RIBS MAY DEVELOP A BOW. DIAMOND-SHAPED L.E. MAY SPLIT RIBS. TISSUE WILL SAG BETWEEN RIBS. BOTTOM SPAR POOR. T.E. GUSSETS ARE GOOD.	REALLY CRUDE! TIP FAR TOO WEAK. IT'S ALL BOUND TO BREAK OR WARP. LAMINATED LEADING EDGE WOULD RELIEVE PRESSURE. SPAR GUSSET WILL PROBABLY FAIL.

## AIRFOIL PROBLEMS:

THERE IS NO PERCENTAGE IN PLOTTING AN AIRFOIL AND THEN MESSENGERING THE JOB UP WITH A SECTION THAT WILL NOT HOLD ITS SHAPE.

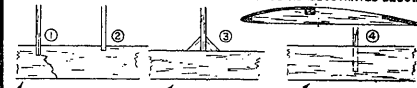


UNDERCAMBER IS FREQUENTLY USED TO ADVANTAGE, BUT BEWARE OF WARPS LIKE THIS. BLAME POOR SPAR-RIB FIT.



## TRAILING EDGE:

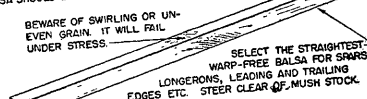
DUE TO THE RELATIVELY SMALL CEMENTING SURFACE OF THE RIB-TRAILING EDGE BUTT-JOINT, IT SHOULD BE STRENGTHENED AS ILLUSTRATED BELOW.



①—GOOD IDEA, BUT TOO DEEP A NOTCH WEAKENS T.E. ② IS EXCELLENT. MAR THE APPEARANCE. ③—TRIANGULAR GUSSETS ARE FINE, BUT APPROACH A DIFFERENT APPROACH. VERY GOOD. ④—SHEET TRAILING EDGE ALLOWS A DIFFERENT APPROACH. VERY GOOD.

## CHOICE OF Balsa:

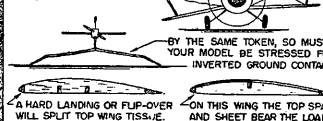
YOUR MODEL, LIKE THE CHAIN WITH A WEAK LINK, IS ONLY AS STRONG AS THE WEAKEST PIECE OF Balsa IN IT. NOTHING BUT PERFECT Balsa SHOULD BE USED FOR MAIN STRUCTURAL MEMBERS.



YOU WILL FIND QUARTER-GRAIN SHEET DIFFICULT TO FLEX OR ROLL. USE THIS RIGIDITY TO ADVANTAGE ON FORMERS, RUDDERS, RIBS AND SIMILAR PLACES. SOFTER MORE FLEXIBLE CUTS OF Balsa NEGOTIATE CURVES OF Balsa EASIER. IDEAL FOR LEADING EDGE PLANKING, FUSELAGE PLANKING, ETC.

## LANDING IMPACT:

A BIPLANE IS STRESSED WITH LANDING WIRES TO PREVENT FOLDING OF WINGS ON HARD LANDINGS, AS AT THE RIGHT.



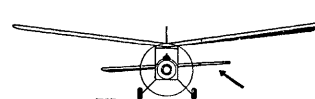
## VIBRATION:

VIBRATION IS THE UNSEEN MODEL KILLER!



ENGINE MUST BE TIGHT ON MOUNTS. FOAM RUBBER MAY BE USED IN MANY CASES TO ABSORB VIBRATION.

## WARPS—ALIGNMENT:



WING AND STAB MUST BE SHIMMED INTO ALIGNMENT IN SUCH CASES. STAB REST MUST BE FIRM. LOCK STAB WITH DOWEL PINS TO PREVENT SIDE MOVEMENT.

REMOVE WARPS WITH HOT WATER OR STEAM.

## REPAIRS:



WHEN POSSIBLE, MAJOR REPAIRS ARE BEST MADE AT HOME. DOUBLE COAT ALL END GRAIN WITH SLOW DRYING MODEL CEMENT.

DO THE JOB RIGHT TO AVOID A RE-REPAIR ON THE FIELD.

Q.J.H.

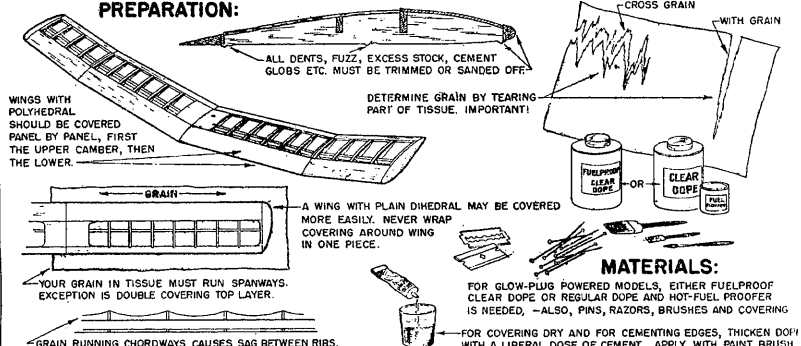
# 14. COVERING & DOPING WINGS

## COVERING MATERIALS:

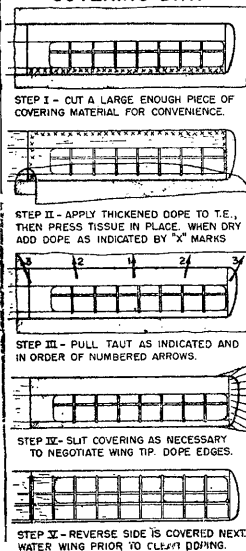
SELECT YOUR COVERING MATERIAL FOR THE JOB WHICH IT MUST DO. FOR THE AVERAGE GAS OR RUBBER MODEL, ANY OF THE STANDARD GRADES OF TISSUE WILL DO NICELY. WHEN ADDED DURABILITY IS DESIRED, USE SILK OR NYLON.

**SILKSPAN:** WHITE ONLY - APPLY WET OR DRY - CAN BE APPLIED WET OVER COMPOUND CURVED MODELSPAN: ALL COLOURS; APPLY DRY - AVAILABLE IN TWO WEIGHTS.  
**JAP TISSUE:** SCARCE AS HEN'S TEETH. LIGHT - REQUIRES LESS DOPE - MUST BE APPLIED DRY.  
**CHINA SILK:** DIE TO DESIRED COLOR - VERY DURABLE - EXCELLENT FOR ALL COMPOUND CURVED.  
**NYLON:** DIE TO DESIRED COLOR - VERY DURABLE, BUT MUST BE PULLED VERY TIGHT WHEN WET.  
**MICROFILM:** REFLECTS SPECTRUM, TRANSPARENT. FANTASTICALLY LIGHT - INDOOR MODELS ONLY.

## PREPARATION:



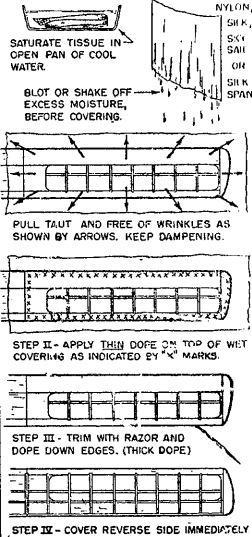
## COVERING DRY:



## DO'S AND DON'TS:

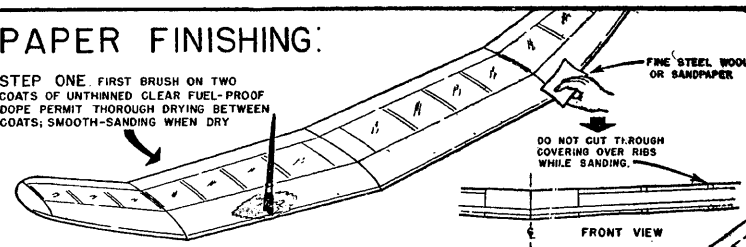
WHEN COVERING A FLAT BOTTOM WING, IT IS NOT ADVISABLE TO DOPE COVERING TO EACH RIB AND SPAR, AS THIS MIGHT CAUSE UNEVEN SHRINKAGE.  
 IF THE BOTTOM OF YOUR WING IS UNDERCAMBERED, APPLY A COAT OF THINNED-DOWN CEMENT TO EACH RIB AND SPAR TO PREVENT THE COVERING FROM BRIDGING THE CONCAVE SURFACE.  
 DO NOT USE SILK ON WEAK OR LIGHTWEIGHT STRUCTURES. HEAVILY DOPED SILK HAS BEEN KNOWN TO CRUSH AND WARP FRAMEWORK.  
 RELY ON COLORED TISSUE OR DYES FOR COLOR ON FREE-FLIGHT MODELS, RATHER THAN EXCESSIVE QUANTITIES OF HEAVY PIGMENTED DOPES.  
 WHEN USING FUELPROOF DOPE AND ALLIED PRODUCTS, DO NOT MIX BRANDS. SANDING SEALERS DESIGNED FOR USE WITH THE SAME COMPOUNDS CLEAR DOPE, MAY CAUSE TROUBLE WHEN APPLIED UNDER A COMPETITIVE BRAND.  
 IF YOUR MODEL IS INTENDED FOR USE AS A SEAPLANE, IT IS ADVISABLE TO SPRAY THE ENTIRE FRAMEWORK WITH TWO COATS OF CLEAR DOPE. THIS WILL RETARD ABSORPTION OF MOISTURE. JAP TISSUE IS PREFERRED FOR COVERING, AS IT HAS FAR LESS PORES AND IS THEREFORE EASIER TO WATERPROOF.  
 DOUBLE-COVER YOUR MODEL IF YOU SEE FIT. A FEW COATS OF CLEAR DOPE SHOULD BE APPLIED TO THE FIRST LAYER. DOPE SECOND LAYER ON TO AVOID AIR BUBBLES. CROSS-GRAIN TISSUE TO LOCALIZE PUNCTURES.  
 DO NOT DOPE YOUR MODEL, IF POSSIBLE, ON WARM HUMID DAYS. THE DOPE WILL TEND TO TURN MILKY-WHITE (BLUSHING). RETARDER WILL PREVENT AS WELL AS REMOVE IT. THINNER MAY HELP TRY TO DOPE IN A COOL DRY ROOM.  
 WHEN COVERING WITH WET TISSUE, THE DOPE APPLIED TO HOLD THE TISSUE WILL BLUSH. IF WEATHER IS DRY, IT WILL DISAPPEAR WHEN DOPED.

## COVERING WET:

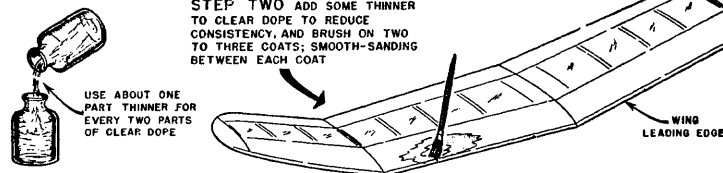


## PAPER FINISHING:

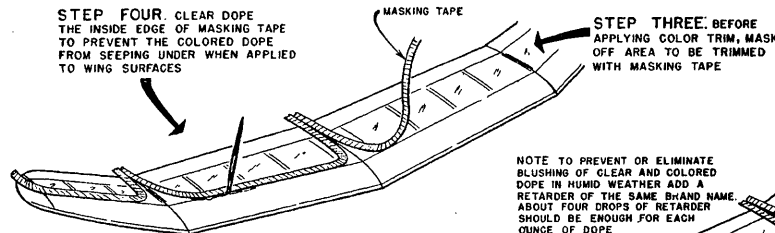
**STEP ONE** FIRST BRUSH ON TWO COATS OF UNTHINNED CLEAR FUEL-PROOF DOPE PERMIT THOROUGH DRYING BETWEEN COATS; SMOOTH-SANDING WHEN DRY



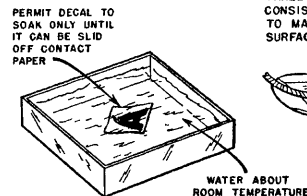
**STEP TWO** ADD SOME THINNER TO CLEAR DOPE TO REDUCE CONSISTENCY AND BRUSH ON TWO TO THREE COATS; SMOOTH-SANDING BETWEEN EACH COAT



**STEP FOUR** CLEAR DOPE THE INSIDE EDGE OF MASKING TAPE TO PREVENT THE COLORED DOPE FROM SEEPING UNDER WHEN APPLIED TO WING SURFACES

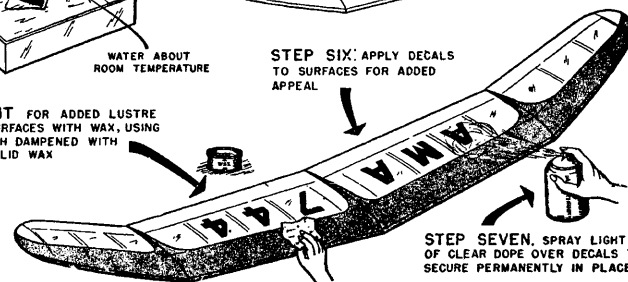


**STEP FIVE** APPLY TWO TO THREE COATS OF MEDIUM CONSISTENCY COLORED DOPE TO MASKED LEADING EDGE SURFACES.



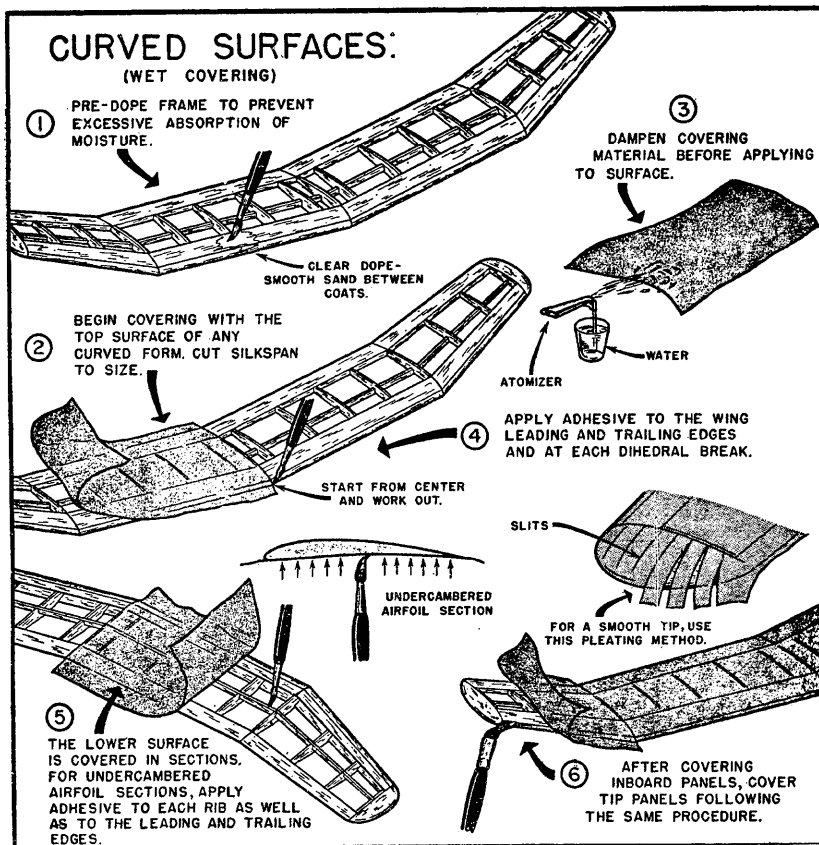
**STEP SIX** APPLY DECALS TO SURFACES FOR ADDED APPEAL

**STEP EIGHT** FOR ADDED LUSTRE RUB DOWN SURFACES WITH WAX, USING A SOFT CLOTH DAMPENED WITH LIQUID OR SOLID WAX



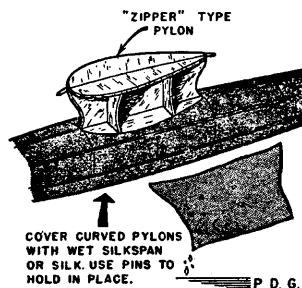
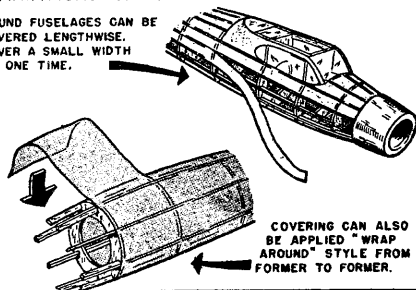
## CURVED SURFACES:

(WET COVERING)



### VARIATIONS ON CURVED SURFACE COVERING:

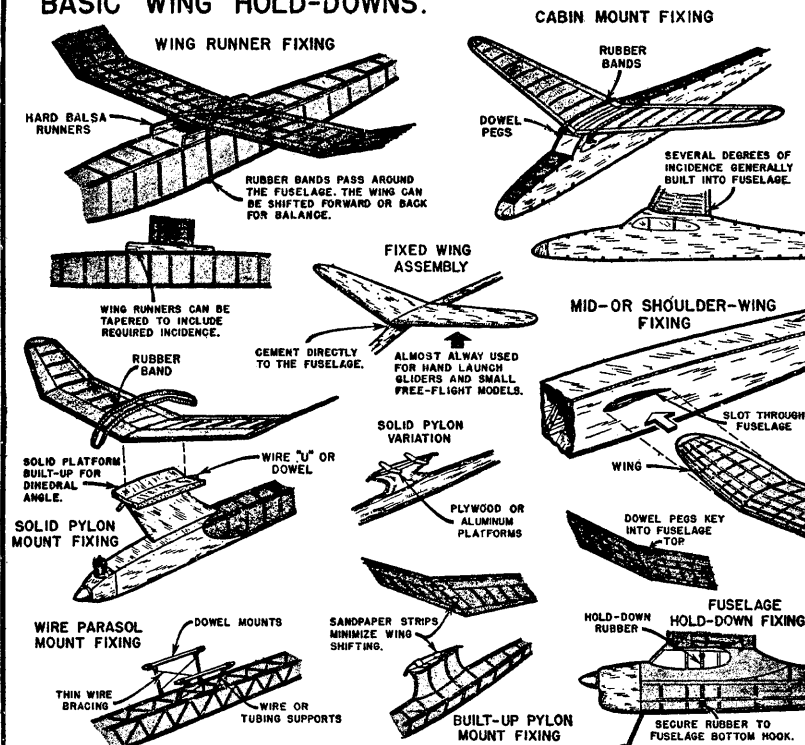
ROUND FUSELAGES CAN BE COVERED LENGTHWISE. COVER A SMALL WIDTH AT ONE TIME.



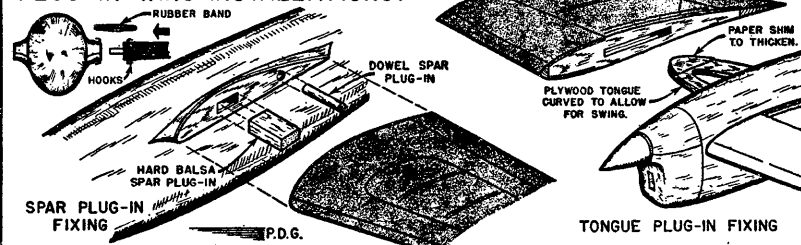
P. D. G.

## 15. WING ATTACHMENT METHODS

### BASIC WING HOLD-DOWNS:

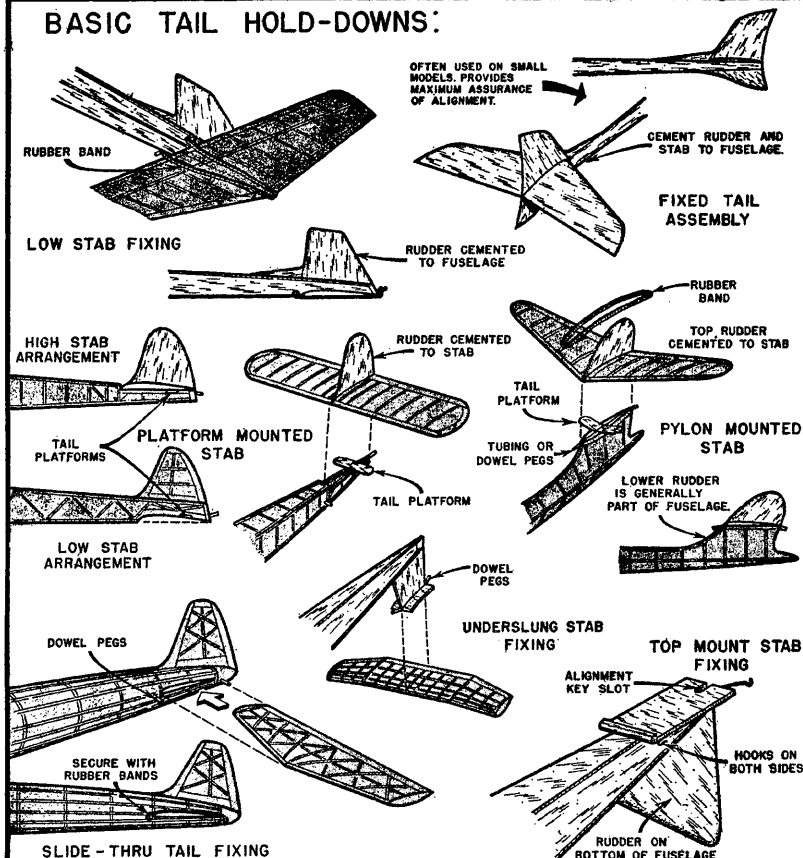


### PLUG-IN WING INSTALLATIONS:

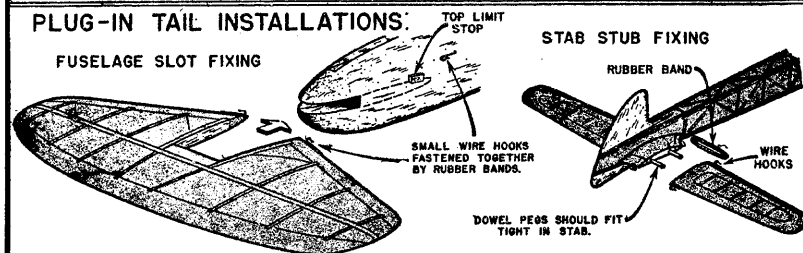


P. D. G.

## BASIC TAIL HOLD-DOWNS:



## PLUG-IN TAIL INSTALLATIONS:

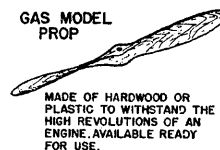


P. D. G

## 16. PROPELLERS FOR ALL PURPOSES

## PROPELLER CLASSIFICATIONS

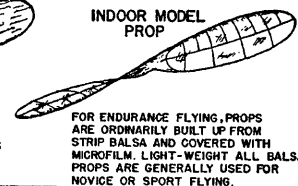
GAS MODEL PROP



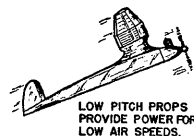
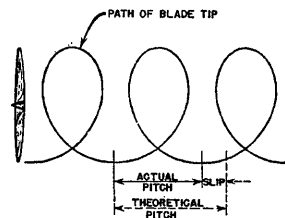
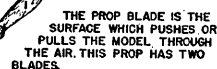
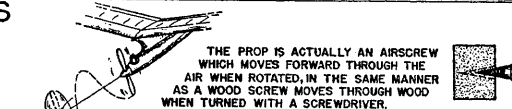
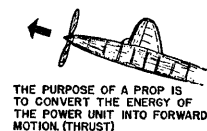
RUBBER MODEL PROP



INDOOR MODEL PROP



## PROPELLER FACTS

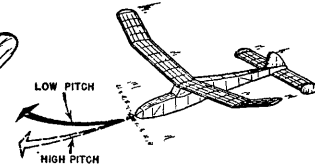


THE DISTANCE A PROP COULD MOVE FORWARD IN ONE REVOLUTION, IF THERE WERE NO RESISTANCE OFFERED BY THE MODEL ("SLIP") IS KNOWN AS THE "THEORETICAL PITCH". THE "ACTUAL PITCH" IS THE TRUE DISTANCE TRAVELED. COMMERCIAL PROPS ARE LISTED ACCORDING TO THEIR THEORETICAL PITCH.

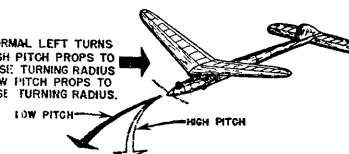
## CHECK FOR BALANCE.



A POORLY BALANCED PROP RESULTS IN POOR PERFORMANCE & PUTS A STRAIN ON THE POWER UNIT. CHECK FOR BALANCE ON A KNIFE EDGE AT CENTER & TRIM HEAVY BLADE, OR ADD FINISH TO LIGHT BLADE, TO OBTAIN PERFECT BALANCE.



FOR NORMAL LEFT TURNS USE HIGH PITCH PROPS TO DECREASE TURNING RADIUS AND LOW PITCH PROPS TO INCREASE TURNING RADIUS.



P.D.G.

## POPULAR PROPELLER TYPES

TWO-BLADED PROPS ARE MOST COMMONLY USED.

THREE-BLADED & FOUR-BLADED PROPS ARE GENERALLY USED FOR SCALE OR SPORT MODELS.

PADDLE-BLADE TYPE PROPS ARE USED ON ALL TYPES OF MODELS, PARTICULARLY RUBBER-POWERED OR CONTROL-LINE STUNT MODELS.

SINGLE-BLADE PROPS ARE MOST OFTEN USED ON HIGH PERFORMANCE RUBBER-POWERED MODELS, BUT HAVE ALSO BEEN USED SUCCESSFULLY ON GAS MODELS.

"SCIMITAR"-SHAPED PROPS ARE CONSIDERED VERY EFFICIENT & HAVE BEEN USED ON ALL TYPES OF MODELS.

## PROPELLER MECHANISMS

THE ROTATION OF THE PROPELLER KEEPS THE BLADES EXTENDED BY CENTRIFUGAL FORCE WHILE POWER IS ON. WITH POWER OFF, THE AIR PUSHES THE HINGED BLADES BACK AGAINST THE FUSELAGE INTO THE PATH OF LEAST RESISTANCE.

FEATHERING PROP

FEATHERED POSITION

WITH POWER ON, THE AIR STRIKING THE BLADES KEEPS THEM AT THE DESIRED PITCH. WITH POWER OFF, THE BLADES PIVOT ON THE HUB LEADING EDGE SO THEY FACE INTO THE PATH OF LEAST RESISTANCE.

FOLDING PROP

FREE-WHEELING PROP

VARIABLE PITCH PROP

THE PROP BLADES ARE MOUNTED SO THAT THEY AUTOMATICALLY ADJUST THEMSELVES, WHILE ROTATING, TO THE PROP PITCH BEST SUITED TO CHANGES IN MODEL'S AIRSPEED, OR ROTATIONAL SPEED OF PROP, TO OBTAIN OPTIMUM EFFICIENCY.

WITH POWER OFF, THE PROP IS DISENGAGED FROM THE ENGINE UNIT. CONTINUED ROTATION IS INDUCED AS A RESULT OF THE AIR STRIKING THE BLADES, CAUSING THEM TO MOVE CONTINUOUSLY IN A PATH OF LOWERED RESISTANCE.

## PROPELLER SELECTION

(RECOMMENDED PROP SIZES, IN INCHES, FOR VARIOUS MODELS)

FREE-FLIGHT GAS						CONTROL-LINE					
LOW PITCH	MED. PITCH	HIGH PITCH	LOW PITCH	MED. PITCH	HIGH PITCH	LOW PITCH	MED. PITCH	HIGH PITCH	LOW PITCH	MED. PITCH	HIGH PITCH
DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH
5-2	5-3	5-4	4-2	4-3	4-4	5-2	5-3	5-5	5-2	5-3	5-5
6-2	6-3	6-4	5-2	5-3	5-5	6-2	6-3	6-4	5-2	5-3	5-5
7-3	7-4	7-5	6-3	6-4	6-6	7-3	7-4	7-5	6-3	6-4	6-6
8-3	8-4	8-5	7-3	7-5	7-7	8-3	8-4	8-5	7-3	7-5	7-7
9-3	9-4	9-6	8-4	8-6	8-8	9-3	9-4	9-6	8-4	8-6	8-8
10-3	10-5	10-6	9-4	9-7	9-9	10-3	10-5	10-6	9-4	9-7	9-9
11-4	11-5	11-7	10-5	10-7	10-10	11-4	11-5	11-7	10-5	10-7	10-10
12-4	12-6	12-8	11-5	11-8	11-11	12-4	12-6	12-8	11-5	11-8	11-11
13-4	13-6	13-9	12-6	12-9	12-12	13-4	13-6	13-9	12-6	12-9	12-12
14-5	14-7	14-10	13-7	13-10	13-13	14-5	14-7	14-10	13-7	13-10	13-13

LIMITED RUBBER						WAKEFIELD					
LOW PITCH	MED. PITCH	HIGH PITCH	LOW PITCH	MED. PITCH	HIGH PITCH	LOW PITCH	MED. PITCH	HIGH PITCH	LOW PITCH	MED. PITCH	HIGH PITCH
DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH
7-6	7-9	7-13	13-13	13-19	13-26	7-6	7-9	7-13	13-13	13-19	13-26
8-7	8-10	8-14	14-14	14-20	14-28	8-7	8-10	8-14	14-14	14-20	14-28
9-7	9-11	9-15	15-15	15-21	15-30	9-7	9-11	9-15	15-15	15-21	15-30
10-8	10-12	10-17	16-16	16-22	16-32	10-8	10-12	10-17	16-16	16-22	16-32
11-8	11-13	11-19	17-17	17-23	17-34	11-8	11-13	11-19	17-17	17-23	17-34
12-9	12-14	12-21	18-18	18-25	18-36	12-9	12-14	12-21	18-18	18-25	18-36
13-10	13-15	13-23	19-19	19-27	19-38	13-10	13-15	13-23	19-19	19-27	19-38
14-11	14-17	14-25	20-20	20-29	20-40	14-11	14-17	14-25	20-20	20-29	20-40
15-12	15-19	15-27	21-21	21-31	21-42	15-12	15-19	15-27	21-21	21-31	21-42
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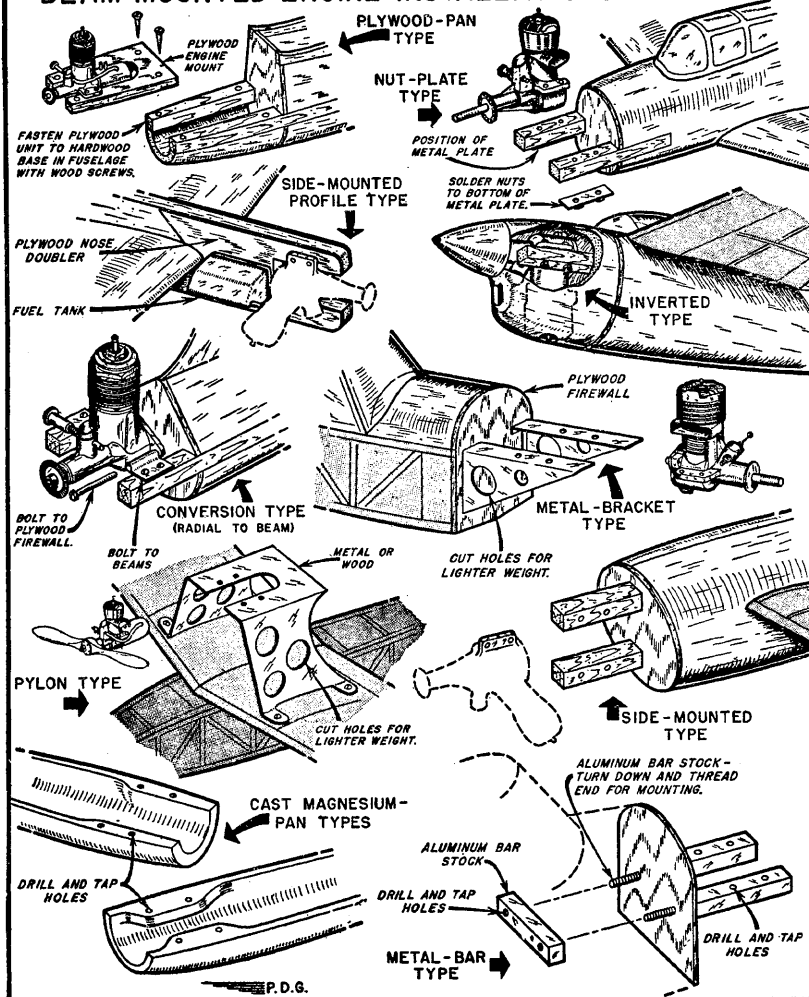
INDOOR STICK						INDOOR CABIN					
LOW PITCH	MED. PITCH	HIGH PITCH	LOW PITCH	MED. PITCH	HIGH PITCH	LOW PITCH	MED. PITCH	HIGH PITCH	LOW PITCH	MED. PITCH	HIGH PITCH
DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH	DIA. PITCH
11-11	11-16	11-22	9-7	9-12	9-17	11-11	11-16	11-22	9-7	9-12	9-17
12-12	12-18	12-24	10-8	10-13	10-18	12-12	12-18	12-24	10-8	10-13	10-18
13-13	13-19	13-26	11-8	11-14	11-19	13-13	13-19	13-26	11-8	11-14	11-19
14-14	14-21	14-28	12-9	12-15	12-20	14-14	14-21	14-28	12-9	12-15	12-20
15-15	15-22	15-30	13-10	13-16	13-22	15-15	15-22	15-30	13-10	13-16	13-22
16-16	16-24	16-32	14-11	14-18	14-24	16-16	16-24	16-32	14-11	14-18	14-24
17-17	17-25	17-34	15-12	15-19	15-26	17-17	17-25	17-34	15-12	15-19	15-26
18-18	18-27	18-36	16-13	16-20	16-28	18-18	18-27	18-36	16-13	16-20	16-28
19-19	19-28	19-38	17-14	17-22	17-30	19-19	19-28	19-38	17-14	17-22	17-30
20-20	20-30	20-40	18-15	18-23	18-32	20-20	20-30	20-40	18-15	18-23	18-32

P.D.G.

THE PITCH SHOWN IN THESE TABLES IS THE THEORETICAL PITCH!

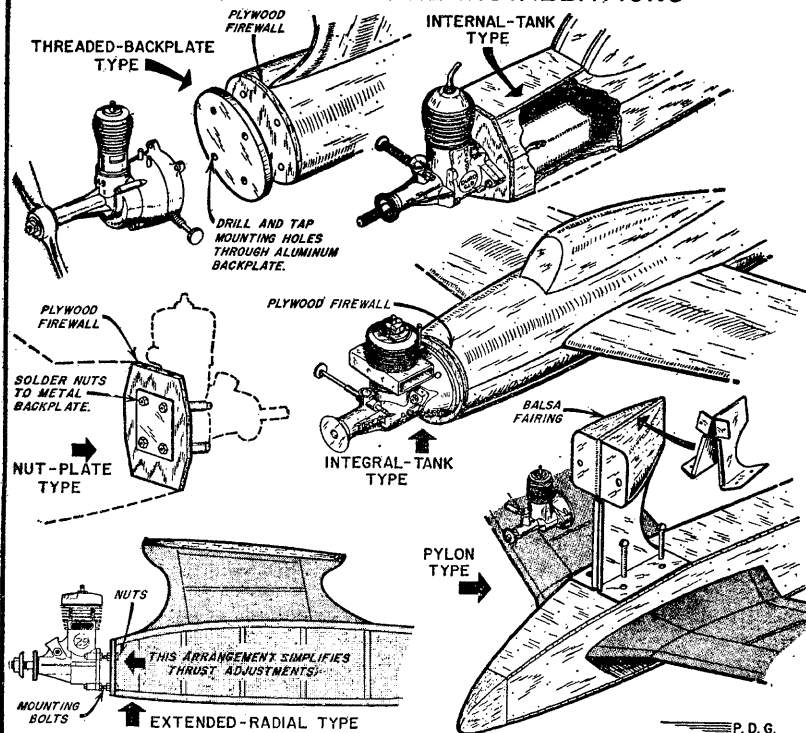
## 17. ENGINE MOUNTINGS

## BEAM MOUNTED ENGINE INSTALLATIONS

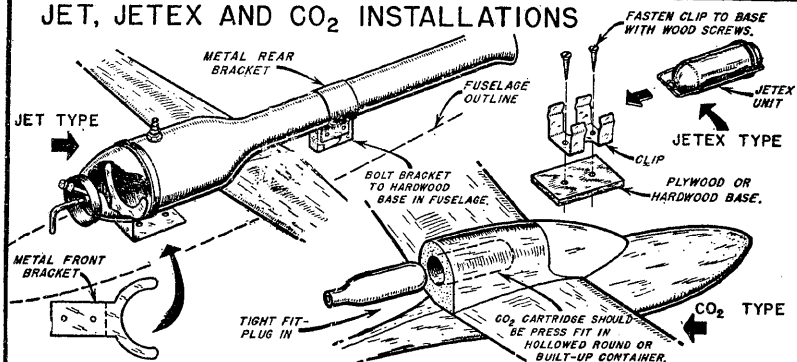




## RADIALLY MOUNTED ENGINE INSTALLATIONS



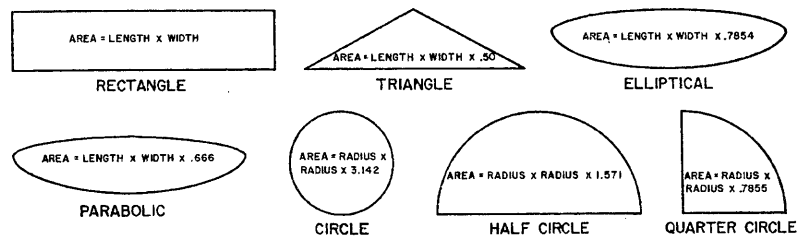
## JET, JETEX AND CO<sub>2</sub> INSTALLATIONS



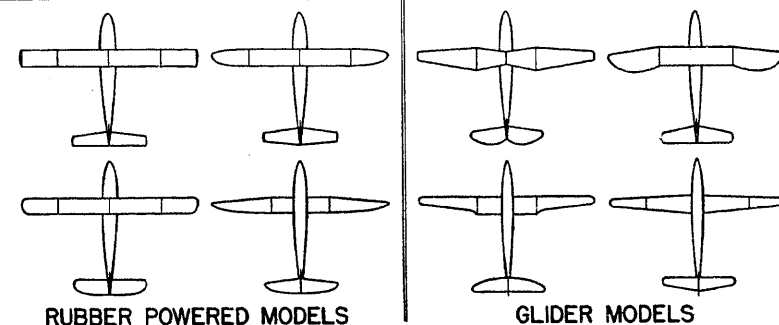
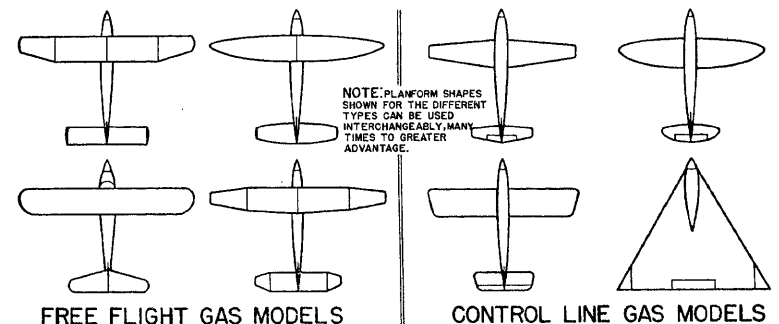
## 18. PLANFORMS IN GENERAL

### WING & STAB PLANFORMS

#### BASIC PLANFORM SHAPES

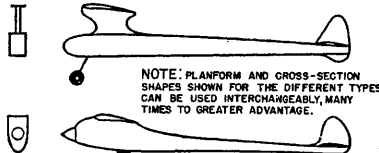


#### GENERAL APPLICATIONS OF BASIC & COMPOSITE PLANFORMS

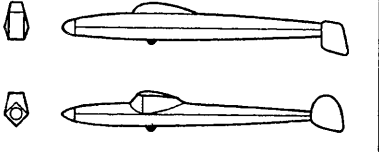


## FREE-FLIGHT

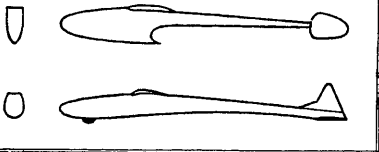
## TYPICAL GAS:



## TYPICAL RUBBER:

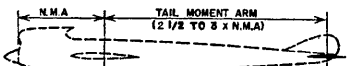


## TYPICAL TOWLINE:



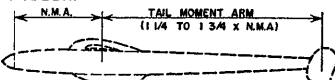
## GENERAL APPLICATION OF MOMENT ARMS:

## SPEED:



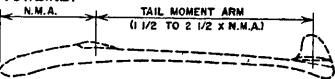
FOR MAXIMUM CONTROLLABILITY, LONG TAIL MOMENT ARM IS DESIRABLE. CENTER OF GRAVITY GENERALLY LOCATED AROUND WING LEADING EDGE DUE TO HIGH CONCENTRATION OF WEIGHT AT THE NOSE. THE USE OF SWEEP FORWARD WING PANELS WILL AID IN OBTAINING A MORE DESIRABLE CENTER OF GRAVITY LOCATION WITH MINIMUM FUSELAGE LENGTH.

## RUBBER:



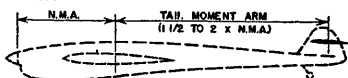
MODERATELY LONG TAIL MOMENT ARM DESIRABLE BUT NOT EASILY OBTAINED BECAUSE OF LENGTH AND WEIGHT OF RUBBER MOTOR. REAR MOTOR BEARING LOCATED WELL FORWARD OF STABILIZING SURFACES ASSISTS IN OBTAINING A MORE DESIRABLE ARRANGEMENT.

## TOWLINE:



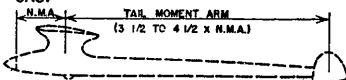
BOTH LONG AND SHORT TAIL MOMENT ARMS USED FOR EASIER TOWING AND TIGHTER GLIDING CIRCLE. SHORT TAIL MOMENT ARM DESIRABLE. LONG TAIL MOMENT ARM DESIRABLE IN GUSTY WEATHER. BALLAST IS USED TO OBTAIN CORRECT CENTER OF GRAVITY LOCATION.

## STUNT:



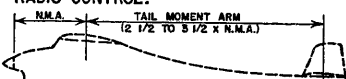
SHORT TAIL MOMENT ARM BEST FOR PROVIDING RAPID CONTROL RESPONSE AND TIGHT BUT SMOOTH MANEUVERS. CENTER OF GRAVITY GENERALLY LOCATED SLIGHTLY FORWARD OF BELLGRANK POSITION.

## GAS:



LONG TAIL MOMENT BEST AND IS EASILY OBTAINED BECAUSE OF HIGH CONCENTRATION OF WEIGHT AT NOSE. CENTER OF GRAVITY POSITION WILL VARY DEPENDING ON THE PARTICULAR DESIGN AND THE PROPORTIONS OF THE SURFACES.

## RADIO CONTROL:

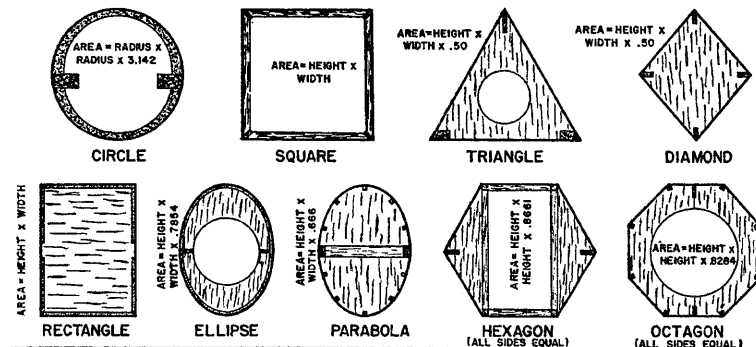


LONG TAIL MOMENT ARM COMMONLY USED, BUT NOT ESSENTIAL. CENTER OF GRAVITY GENERALLY LOCATED 1/4 TO 1/3 FROM THE WING LEADING EDGE - EASY TO OBTAIN BEST POSITION BY SHIFTING LOCATION OF THE RADIO EQUIPMENT

## 19. CONTROL LINE PLANFORMS

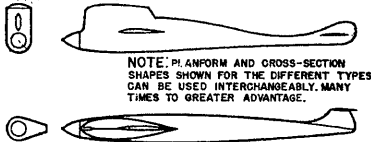
## FUSELAGE AND RUDDER PLANFORMS

## BASIC FUSELAGE CROSS-SECTION SHAPES:

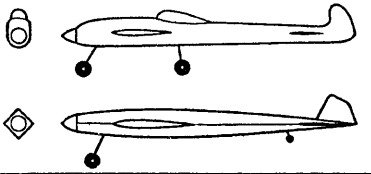


## GENERAL APPLICATIONS OF BASIC AND COMPOSITE PLANFORMS CONTROL-LINE

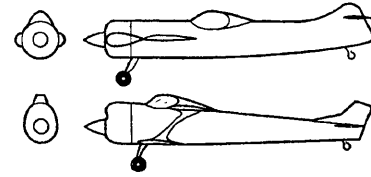
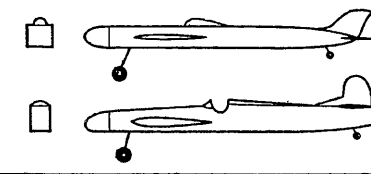
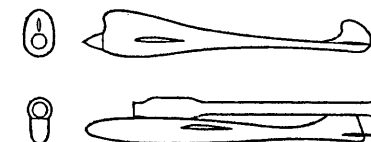
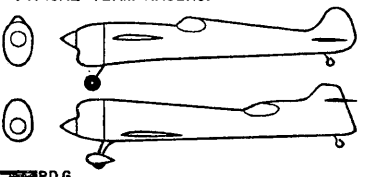
## TYPICAL SPEED:



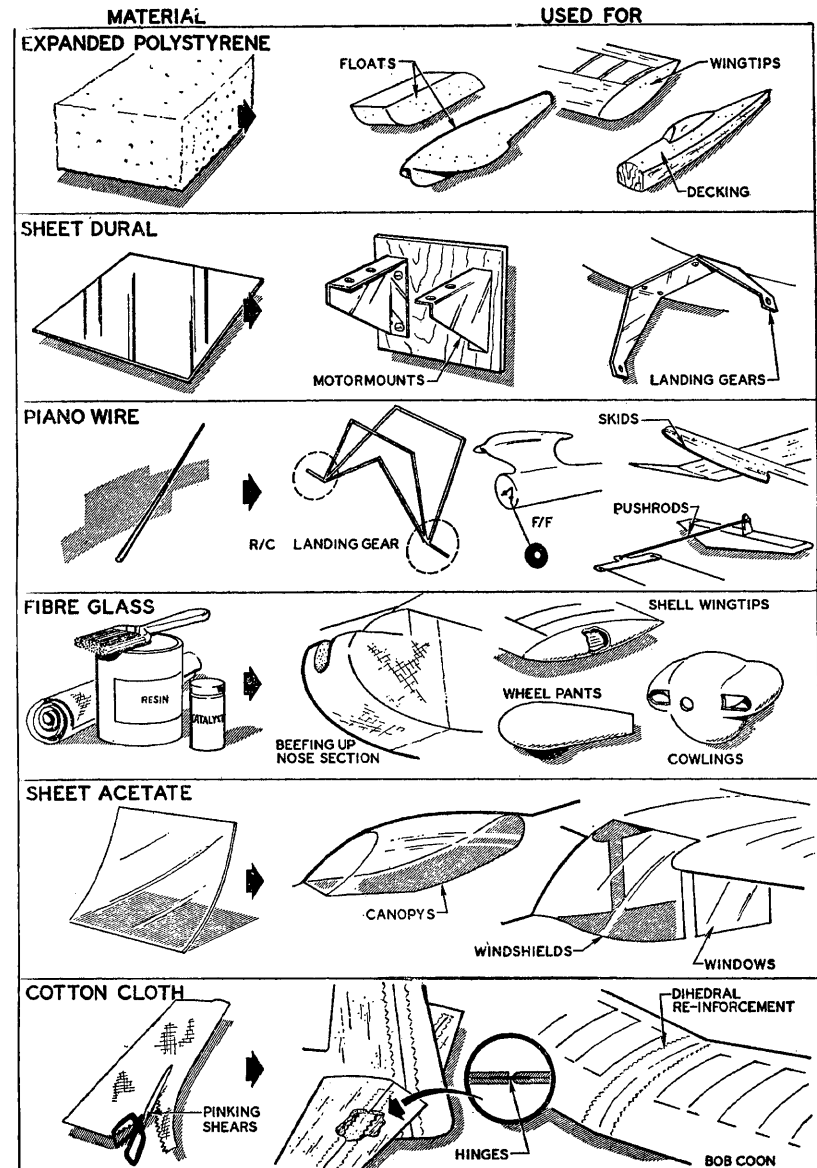
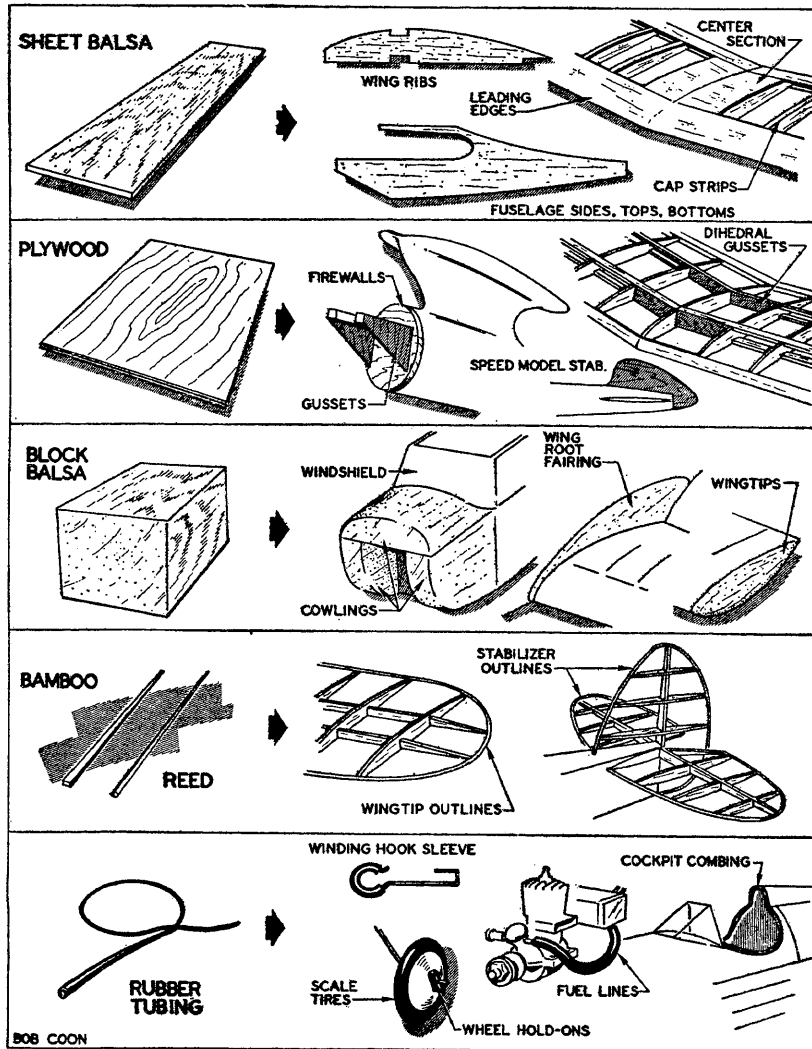
## TYPICAL STUNT:



## TYPICAL TEAM RACERS:



## 20. USE OF MATERIALS



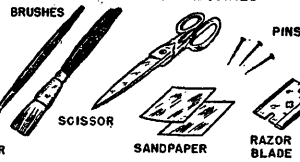
## 21. FUSELAGE COVERING PROCEDURE

### FLAT SURFACES: (DRY COVERING)

USE A 50-50 MIXTURE AS AN ADHESIVE.



#### MATERIALS REQUIRED



TYPICAL BUILT-UP FUSELAGE

COVERING MATERIALS

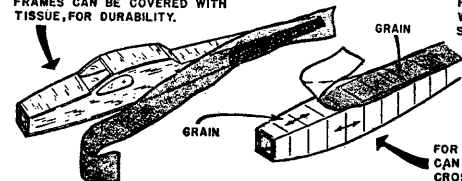
### COVERING WITH TISSUE PAPER

1. PRESS OUT ALL WRINKLES FROM TISSUE PAPER WITH WARM IRON.
2. CUT STRIP FOR ALL OR PART OF ONE SIDE, DEPENDING ON LENGTH.
3. APPLY ADHESIVE ONLY TO LONGERONS AND DIAGONAL MEMBERS AT EACH END OF FRAME, OF SIDE BEING COVERED.
4. TRIM EXCESS WITH RAZOR BLADE.
5. TOP MUST BE COVERED IN SECTIONS AROUND THE PYLON.
6. DAB TRIMMED EDGES WITH COVERING ADHESIVE, WHERE NOT SECURELY ADHERED, AND SMOOTH OUT WITH FINGERS.
7. SPRAY COVERING WITH WATER, UNTIL THE COVERING IS COMPLETELY DAMPENED. DO NOT SOAK COVERING!

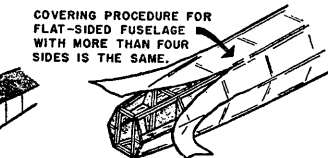
LARGER MODELS CAN BE COVERED WITH WET SILK, SILKSPAN, OR SKYSAIL, BUT NOT WITH TISSUE BECAUSE IT FALLS APART.

### VARIATIONS ON FLAT SURFACE COVERING

COMPLETELY WOOD COVERED FRAMES CAN BE COVERED WITH TISSUE, FOR DURABILITY.



COVERING PROCEDURE FOR FLAT-SIDED FUSELAGE WITH MORE THAN FOUR SIDES IS THE SAME.



FOR ADDED STRENGTH, MODELS CAN BE DOUBLE-COVERED. CROSS-GRAIN SECOND LAYER.

P. D. G.

ALL COVERING MATERIAL, WHETHER SILK OR TISSUE, IS ESSENTIALLY AS FLAT AS A PANCAKE.

IT MAY BE ROLLED INTO A TUBULAR SHAPE QUITE EASILY, WET OR DRY.

HOWEVER, IT CAN NOT BE WRAPPED AROUND A BALL (WHEN DRY) WITHOUT WRINKLING.

BUT, HALF A BALL IS HALF THE PROBLEM.

-AND IF WE COVER WITH WET PAPER, THEN THE JOB WILL BE EASY.

WETTING SILKSPAN, SKYSAIL OR SILK ALLOWS THE FIBRES UNDER PRESSURE TO STRETCH. PULL TO SHAPE SLOWLY, CAREFULLY.

### COMPOUND CURVATURES:

AS DRY TISSUE IS NOT SUITABLE FOR COMPOUND CURVES, WING TIPS SUCH AS THIS SHOULD BE COVERED WET.

THE TAPERING AIRFOIL CREATES A CURVE IN THE THIRD DIMENSION.

NOTE CROSS-SECTION WITH SEGMENTS A, B, C.

SEG. A-INDICATES AREA WHICH MAY BE COVERED WITH WET TISSUE.

SEG. B-MAY BE COVERED WITH ONE PIECE OF SILK.

SEG. C-DRY TISSUE.

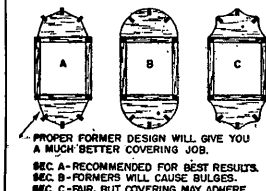
COMPOUND CURVATURES ARE OFTEN FOUND AFT OF THE CABIN, AND MAY ALSO BE COVERED BY THE METHODS DESCRIBED ABOVE.

PARACHUTE IS A PERFECT EXAMPLE OF THE INDIVIDUAL 'GORE' METHOD OF COVERING.

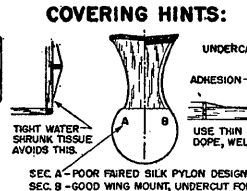
A ROUNDED STRINGERED FUSELAGE MAY BE COVERED DRY, BY COVERING EACH SECTION BETWEEN STRINGERS INDIVIDUALLY.

PAIRED SILK PYLONS ARE DIFFICULT. USE WET SILK ONLY. PULL VERY TIGHT IN ALL DIRECTIONS. HOLD WITH PINS, CEMENT. FORMERS DO NOT TOUCH SILK!

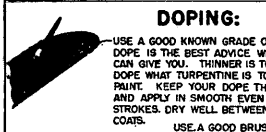
### COVERING HINTS:



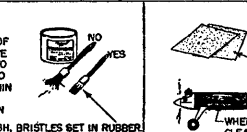
UNDERCAMBER COVERING SHOULD BE CEMENTED TO EACH RIB.



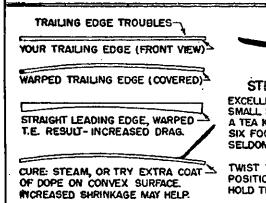
ADHESION: DOPING CAUSES SAGGING OF ROCKY BRACED LONGERONS. USE THIN COATS OF DOPE, WELL BRUSHED.



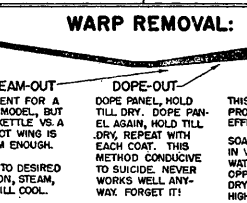
PROPER FORMER DESIGN WILL GIVE YOU A MUCH BETTER COVERING JOB.



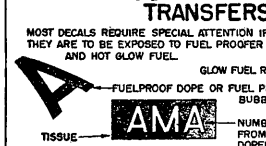
TIGHT WATER SHRINK TISSUE AVOIDS THIS.



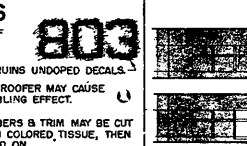
SEC. A-RECOMMENDED FOR BEST RESULTS. SEC. B-FORMERS WILL CAUSE BULGES. SEC. C-PAIR, BUT COVERINGS MAY ADHERE.



SEC. A-POOR FAIRED SILK PYLON DESIGN. SEC. B-GOOD WING MOUNT, UNDERCUT FORMER.



STEAM-OUT: EXCELLENT FOR A SMALL MODEL, BUT A TEA KETTLE VS. A SIX FOOT WING IS SELDOM ENOUGH.



DOPE-OUT: DOPE PANEL, HOLD TILL DRY. DOPE PANEL AGAIN, HOLD TILL DRY. REPEAT WITH EACH COAT. THIS METHOD CONDUCTIVE TO SUICIDE. NEVER WORKS WELL ANYWAY. FORGET IT!



SOAK-OUT: THIS METHOD WILL PROVE VERY EFFECTIVE. SOAK WARPED PANEL IN VERY HOT RAGUET WATER. TWIST IN OPPOSITE DIRECTION. DRY FAST OVER HEAT. HIGHLY INFLAMMABLE.



WARP REMOVAL: OCCASIONALLY RUBBER MODEL FUSELAGES DEVELOP A SLIGHT TWIST BEFORE COVERING. SEE CROSS-SEC.



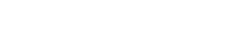
TRANSFERS: MOST DECALS REQUIRE SPECIAL ATTENTION IF THEY ARE TO BE EXPOSED TO FUEL PROOF AND HOT GLOW FUEL.



GLOW FUEL RUINS UNDOPED DECALS. FUELPROOF DOPE OR FUEL PROOFER MAY CAUSE BUBBLING EFFECT.



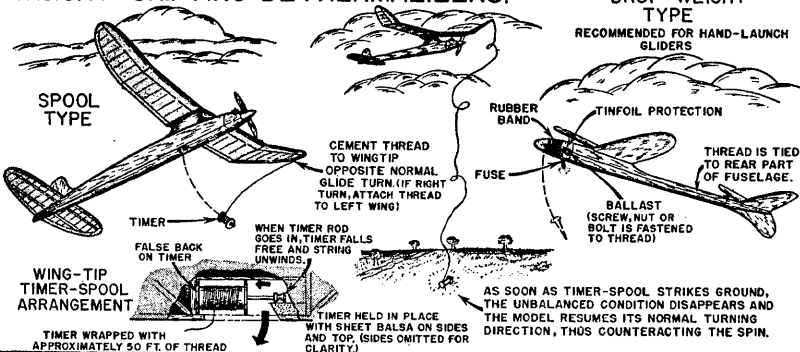
NUMBERS & TRIM MAY BE CUT FROM COLORED, TISSUE, THEN DOPED ON.



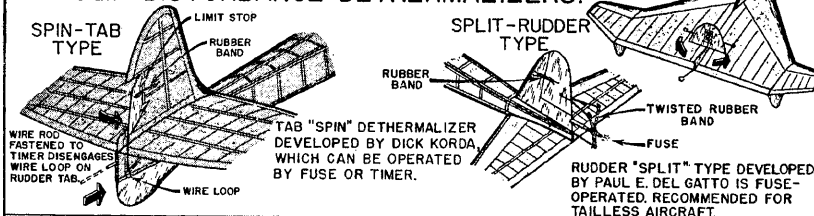
PATCHING: REMOVE RIPPED COVERING WITH A RAZOR. WHENEVER POSSIBLE REMOVE A BIT OF SURROUNDING TISSUE SO PATCH MAY BE CEMENTED TO WOOD, ELIMINATING DOUBLE-COVERING EFFECT. WATER AND DOPE TO MATCH.

## 22. ALL ABOUT DETHERMALISERS

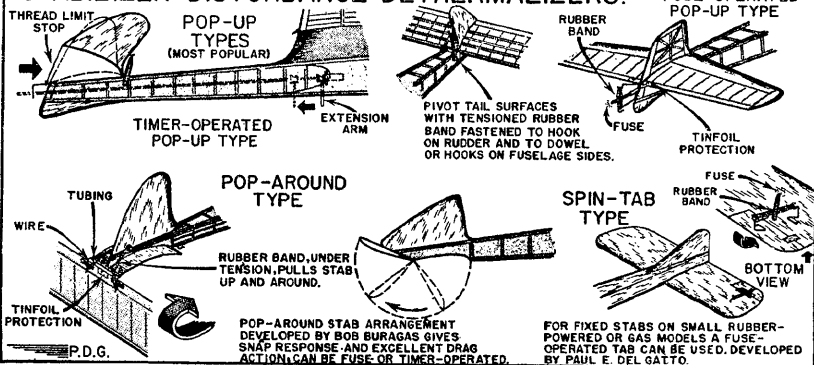
### WEIGHT-SHIFTING DETHERMALISERS:



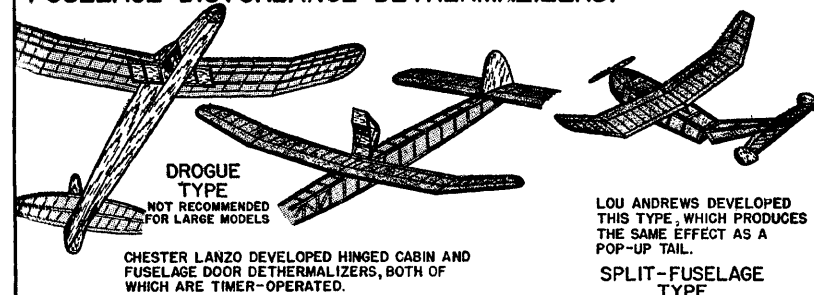
### RUDDER-DISTURBANCE DETHERMALISERS:



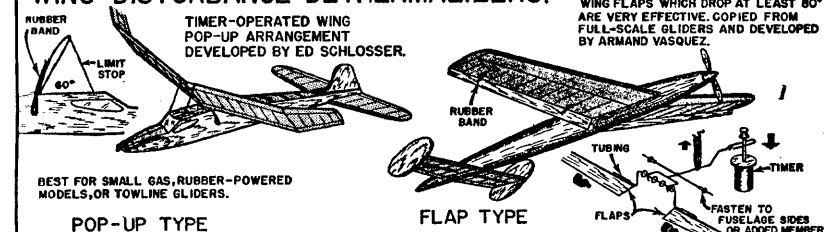
### STABILIZER-DISTURBANCE DETHERMALISERS:



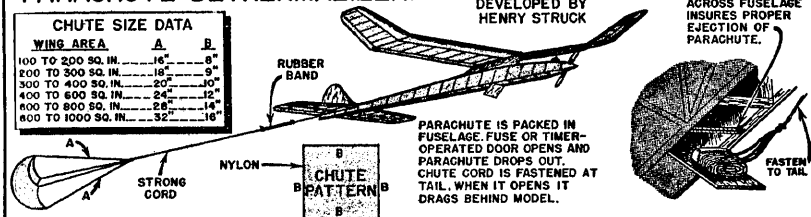
### FUSELAGE-DISTURBANCE DETHERMALISERS:



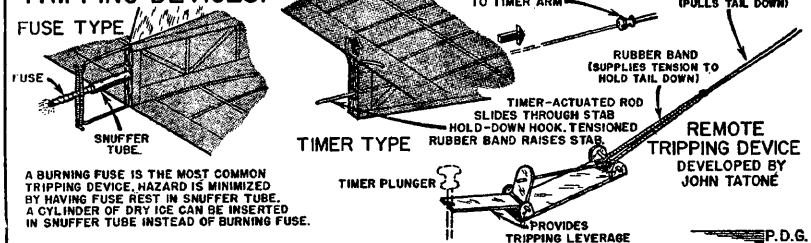
### WING-DISTURBANCE DETHERMALISERS:



### PARACHUTE DETHERMALIZER:



### TRIPPING DEVICES:

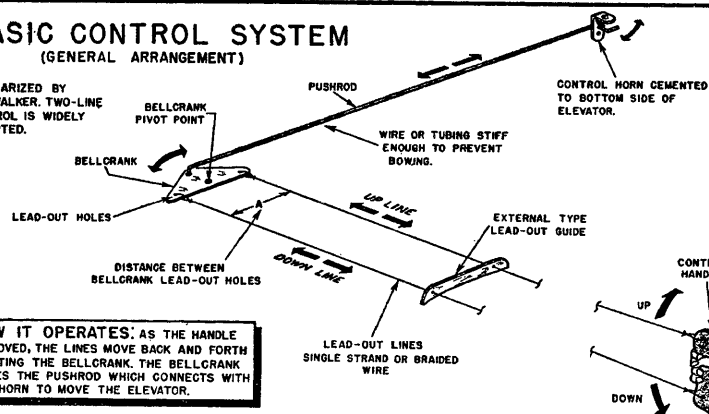




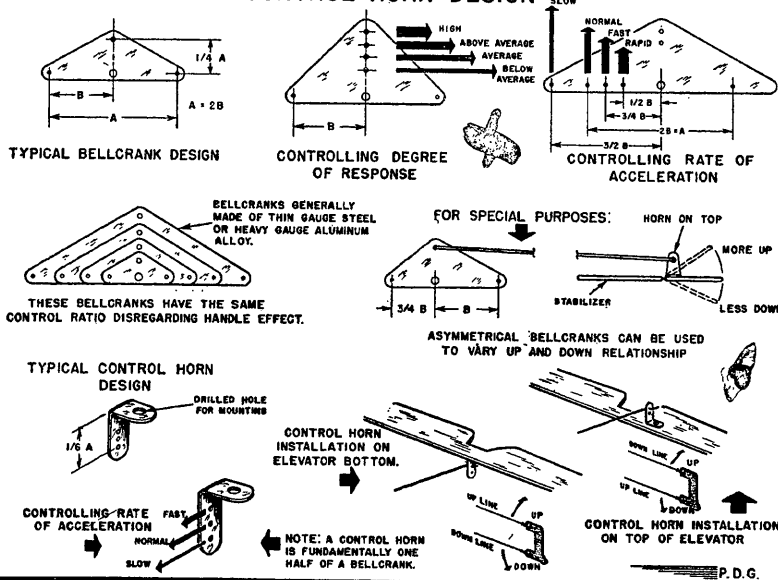
## 23. CONTROL LINE EXPLAINED

### BASIC CONTROL SYSTEM (GENERAL ARRANGEMENT)

POPULARIZED BY  
JIM WALKER, TWO-LINE  
CONTROL IS WIDELY  
ACCEPTED.



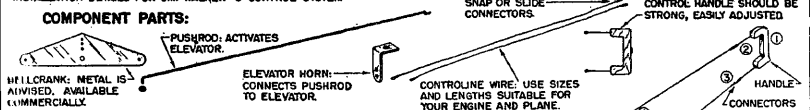
### BELLCRANK AND CONTROL HORN DESIGN



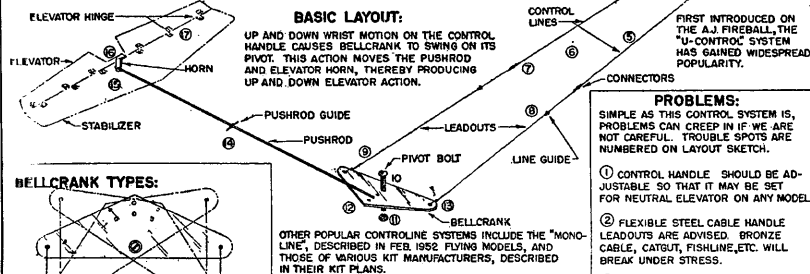
## 24. BASIC CONTROL LINE

INSTALLATION DETAILS FOR JIM WALKER "U-CONTROL" SYSTEM

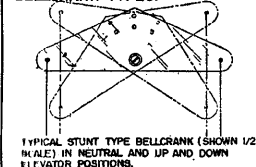
### COMPONENT PARTS:



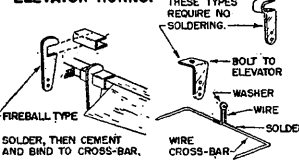
### BASIC LAYOUT:



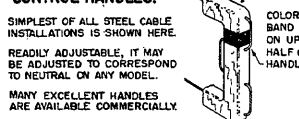
### BELLCRANK TYPES:



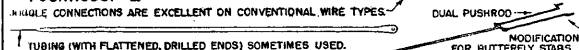
### ELEVATOR HORNS:



### CONTROL HANDLES:

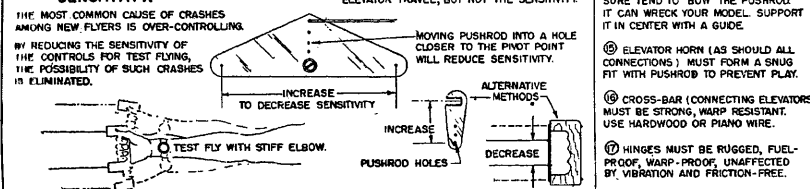


### PUSHRODS:



### SENSITIVITY:

THE MOST COMMON CAUSE OF CRASHES AMONG NEW FLYERS IS OVER-CONTROLLING. BY REDUCING THE SENSITIVITY OF THE CONTROLS FOR TEST FLYING, THE POSSIBILITY OF SUCH CRASHES IS ELIMINATED.



- PROBLEMS:**  
SIMPLE AS THIS CONTROL SYSTEM IS, PROBLEMS CAN CREEP IN IF WE ARE NOT CAREFUL. TROUBLE SPOTS ARE NUMBERED ON LAYOUT SKETCH.
- CONTROL HANDLE SHOULD BE ADJUSTABLE SO THAT IT MAY BE SET FOR NEUTRAL ELEVATOR ON ANY MODEL.
  - FLEXIBLE STEEL CABLE HANDLE LEADOUTS ARE ADVISED. BRONZE CABLE, CATWIP, FISHLINE, ETC. WILL BREAK UNDER STRESS.
  - WEAK CONNECTORS ARE DANGEROUS. PULL TEST TO AVOID ACCIDENTS.
  - DO NOT USE THINNER WIRE THAN RECOMMENDED FOR YOUR ENGINE.
  - LINES MUST BE FREE OF KINKS.
  - LINES SHOULD BE CORRECT LENGTH.
  - ALL WIRE CONNECTIONS SHOULD BE SLIP PROOF, SOLDERLESS, OR SOLDERED.
  - LINE GUIDE ON WING MUST NOT HIT CONNECTORS, OR CHAFE ON LINES. NO LINE GUIDE IS NEEDED IF LINES RUN THROUGH WING.
  - WIRE CONNECTION AT BELLCRANK MUST NOT SNAG ON FUSELAGE SIDE, OR LIMIT BELLCRANK MOVEMENT.
  - IN LARGE OR HIGH POWERED MODELS, A BEARING IS ADVISED AT THE PIVOT. ANCHOR PIVOT BOLT MOUNT TO MOTOR MOUNTS FOR SAFETY.
  - SOLDER NUT TO PREVENT LOOSENING.
  - IF LESS MOVEMENT OF THE ELEVATORS IS DESIRED, USE PUSHROD TRAINING HOLES (NEARER PIVOT) IN BELLCRANK.
  - USE LARGEST BELLCRANK POSSIBLE.
  - FRICTION COUPLED WITH AIR PRESSURE TEND TO "BOW" THE PUSHROD. IT CAN WRECK YOUR MODEL. SUPPORT IT IN CENTER WITH A GUIDE.
  - ELEVATOR HORN (AS SHOULD ALL CONNECTIONS) MUST FORM A SNUG FIT WITH PUSHROD TO PREVENT PLAY.
  - CROSS-BAR (CONNECTING ELEVATORS) MUST BE STRONG, WARP RESISTANT. USE HARDWOOD OR PIANO WIRE.
  - HINGES MUST BE RUGGED, FUEL-PROOF, WARP-PROOF, UNAFFECTED BY VIBRATION AND FRICTION-FREE.

**ELEVATOR HINGES:**

THE CRINOLINE CLOTH HINGE IS EASY TO MAKE, SHORT LIVED.



DOPE AND GLUE ON FLEXIBLE EDGES MAY CRACK THE CLOTH. THE CLOTH ALSO LEAVES A BUMPY EFFECT, MARRING APPEARANCE OF THE TAIL ASSEMBLY.

THE INTERLOCKING WIRE HINGE IS EXCELLENT FOR SCALE JOBS.



HINGES SUCH AS THESE, AVAILABLE AT YOUR HOBBY SHOP ARE EASY TO INSTALL, TROUBLE FREE.

TUBING AND WIRE HINGE



SLIDE WIRE THROUGH TUBING

LENGTHS OF TUBING CEMENTED TO STAB AND ELEVATOR MAKE AN EXCELLENT HINGE.

COVER WITH BALSA, SAND.

TUBING

FILL-IN, SAND

ELEVATOR

**ELEVATOR MOVEMENT:**

FOR TRAINING PURPOSES LIMIT YOUR ELEVATOR MOVEMENT TO 5° DOWN, 15° UP.

LESS DOWN IS NEEDED DUE TO GRAVITY.

STUNT MODELS MAY USE UP TO 45° UP AND DOWN MOVEMENT

PULL UP AND FULL DOWN

IF ELEVATOR IS STILL IN UP POSITION WHEN DOWN ELEVATOR IS DESIRED, SLIDE STAB TOWARD REAR.

CEMENT STAB IN POSITION ONCE PROPER MOVEMENT IS OBTAINED.

**CONNECTIONS:**

BIND AND SOLDER

CONNECTIONS SHOULD BE STRONG, PULL TESTED.

ONE TUG ON THE LINES AND THIS CONNECTION WILL BREAK.

① BEND CONTROL WIRE, SUP SNAP OR SLIDE CONNECTOR ON.

② BIND AS SHOWN WITH SINGLE STRAND OF ELECTRICAL WIRE.

③ BEND LEG OF WIRE DOWN, CONTINUE BINDING. DO NOT SOLDER.

①

②

③

**BELLCRANK MOUNT:**

YOUR MODEL IS A POTENTIAL WRECK, IF ITS BELLCRANK IS NOT RIGIDLY MOUNTED.

IF YOUR BELLCRANK IS ATTACHED TO ANYTHING BUT THE MOUNTS, IT MAY SUDDENLY TEAR OUT.

SCREW PLYWOOD BELLCRANK MOUNT TO MOTOR MOUNTS.

BELLCRANK C.G.

BELLCRANK - 25% FROM L.E. C.G. - 33% FROM L.E.

IF YOUR BELLCRANK IS ATTACHED TO ANYTHING BUT THE MOUNTS, IT MAY SUDDENLY TEAR OUT.

SCREW PLYWOOD BELLCRANK MOUNT TO MOTOR MOUNTS.

SCREW PLYWOOD BELLCRANK MOUNT TO MOTOR MOUNTS.

**LINE GUIDE:**

RUNNING CONTROL LINES THROUGH THE WING WHENEVER POSSIBLE, LESSENS DRAG AND ELIMINATES THE NEED FOR A LINE GUIDE.

NO LINE GUIDE IS NEEDED ON THIS WING.

NO LINE GUIDE IS NECESSARY

LINE GUIDE LOCATION FOR BIPLANE

NOTE LINE GUIDE

**FLIGHT TRIM:**

IMPROPER TRIM ACCOUNTS FOR MANY CRASHES. THE MODEL MUST HOLD TIGHT ON THE LINES AND FLY WITH ITS WINGS PARALLEL TO THE CONTROL LINES AT ALL TIMES.

IF A CONTROL LINE MODEL IS ALLOWED TO BANK, IT WILL LAND AND TAKE-OFF ON ONE WHEEL, TEND TO FLY INSIDE CIRCLE, SLACKENING LINES.

BANKING MAY BE TRIMMED BY WARPING THE TRAILING EDGE OF THE INBOARD PANEL DOWN (WASH-IN) AND THE TRAILING EDGE OF THE OUTBOARD PANEL UP (WASH-OUT)

STANDARD ADJUSTMENTS ARE NUMBERED BELOW.

① SLIGHT RIGHT RUDDER

② SWEEPBACK CONTROL LINES

③ LEAD WEIGHT IN OUTBOARD TIP

④ RIGHT THRUST WHEN NECESSARY.

FIXED ALERONS ARE EXCELLENT FOR TRIMMING BANKING TENDENCIES. TACK LIGHTLY WITH CEMENT, UNTIL FINAL ADJUSTMENTS ARE MADE. ALERONS REPLACE WING WARPING METHOD AT LEFT.

ELEVATORS SHOULD BE PROPERLY ALIGNED ON THE CROSS-BAR, TO PREVENT BANKING.

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**REEL AND LINES:**

CUT ONE 6" DIA. 3/8" PLYWOOD DISC

CUT TWO 7" DIA. 1/8" PLYWOOD DISCS.

DRILL 1/4" HOLE THROUGH DISC CENTERS. CEMENT TOGETHER, ALIGNING HOLES.

NOTCH AT ANGLE.

CRANKING KNOB

SCREW

WASHER

6" DIA. 3/8" PLYWOOD DISC

WASHER

WASHER

CRANKING KNOB

SCREW

WASHER

6" DIA. 3/8" PLYWOOD DISC

WASHER

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CRANKING KNOB

SCREW

WASHER

6" DIA. 3/8" PLYWOOD DISC

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CRANKING KNOB

SCREW

WASHER

6" DIA. 3/8" PLYWOOD DISC

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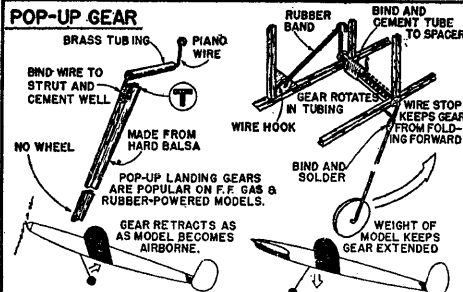
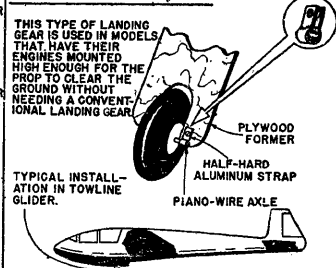
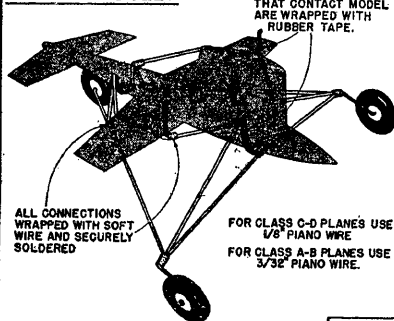
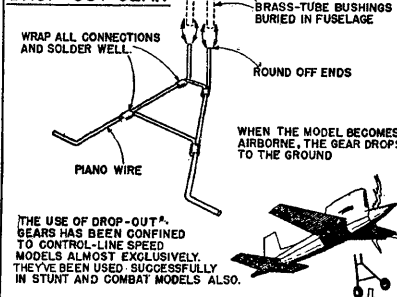
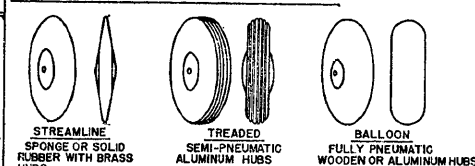
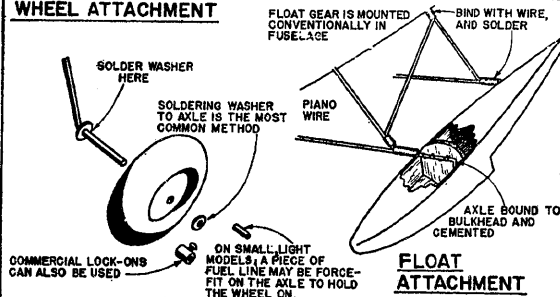
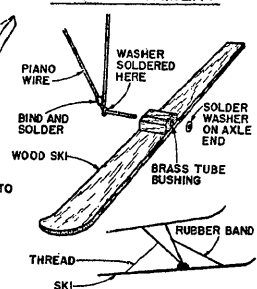
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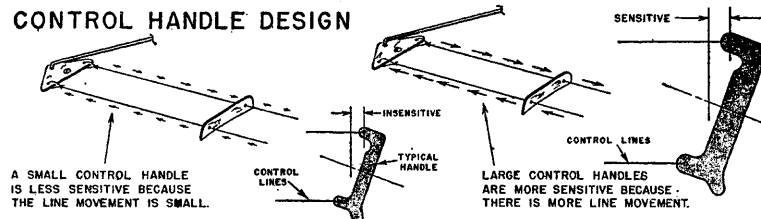
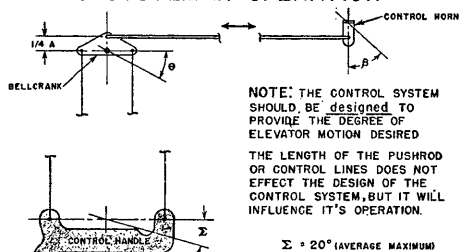
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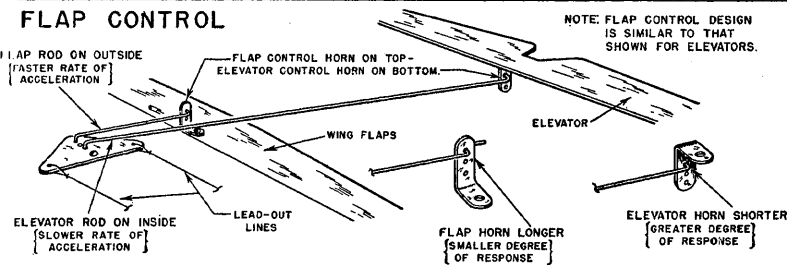
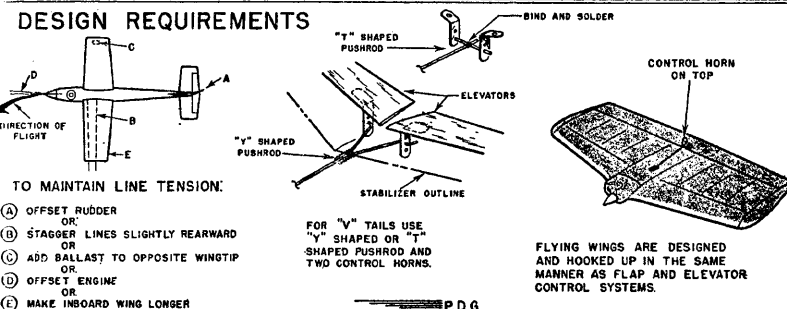
WASHER

**POP-UP GEAR****SINGLE-WHEEL SKID****TAKE-OFF DOLLY****DROP-OUT GEAR****TYPES OF RUBBER-TIRED WHEELS****WHEEL ATTACHMENT****SKI ATTACHMENT****FLOAT ATTACHMENT**

BY HAROLD STEVENSON

**26. C/L HANDLES & OPERATION****CONTROL HANDLE DESIGN****BASIC SYSTEM IN OPERATION****EXAMPLES OF DESIGN**

BELL-CRANK	CONTROL HANDLE	CONTROL HORN	Σ	θ	β
3"	5 1/2"	3/4"	20°	37°	33°
3"	5"	5/8"	20°	34°	36°
3"	4-1/2"	1/2"	20°	30°	41°
2"	5"	1/2"	20°	50°	45°
2"	4-1/2"	1/2"	20°	48°	40°
2"	4"	3/8"	20°	40°	46°
1-1/2"	5"	1/2"	20°	70°	50°
1-1/2"	4-1/2"	3/8"	20°	64°	54°
1-1/2"	4"	5/16"	20°	57°	61°

**FLAP CONTROL****DESIGN REQUIREMENTS**

## 27. TEAM RACING & SPEED FLYING

F.A.I. & S.M.A.E. LINE LENGTHS

CLASS	DISPLACEMENT	LINE LENGTH	DISTANCE	NO. OF LAPS
1/2A 1/2R	0-1.5 CC	38' 2 1/2"	5 MILES	10
1 1/2 1/2R	0-2.5 CC	1592 M (52' 4 1/2")	10 KILOMETRE	10 (1/2R 100)
II	2.5-3.0 CC	1592 M (52' 4 1/2")	10 KILOMETRE	10
III JET	3.1-10.0 CC	1592 M (52' 4 1/2")	10 KILOMETRE	10
SMAE 1/2R B	0-5 CC	60'	5 1/2 MILES	70 & 140

### GENERAL INFORMATION

A RECORD-BREAKING SPEED RUN IS NOT DUE TO A HOPPED-UP ENGINE OR SUPER-SECRET FUEL, AS IS GENERALLY BELIEVED. ATTENTION TO DETAIL IN ALL PHASES OF SPEED FLYING IS THE KEY TO A REALLY FAST FLIGHT. A SO-CALLED HOT FUEL RUNNING IN A DIRTY ENGINE OR A BEAUTIFULLY CONSTRUCTED & FINISHED PLANE WITH A POOR FUEL SYSTEM NEVER WINS CONTESTS. GOOD SPEED FLYING IS THE RESULT OF A CHAIN OF WELL-THOUGHT-OUT DETAILS CAREFULLY CARRIED OUT & YOU ARE ONLY AS FAST AS THE WEAKEST LINK IN THAT CHAIN.

THESE DETAILS ARE AS FOLLOWS:

- 1- A WELL-CONSTRUCTED PLANE OF ADVANCED DESIGN, KIT OR ORIGINAL
- 2- A GLASS-LIKE FUEL-PROOF FINISH
- 3- A PROVEN RACING ENGINE WELL TAKEN CARE OF
- 4- A RELIABLE CONSISTENT FUEL
- 5- A FOOL-PROOF FUEL-SYSTEM (TANK)
- 6- A THOROUGHLY FLIGHT TESTED PROPELLER
- 7- A RELIABLE TAKE-OFF DOLLY OR HAND LAUNCH
- 8- A SET OF A.M.A. SPECIFIED CONTROL LINES
- 9- CLEANLINESS - THIS APPLIES TO ALL EIGHT POINTS MADE ABOVE
- 10- PLENTY OF PRACTICE FLYING ON THE PYLON

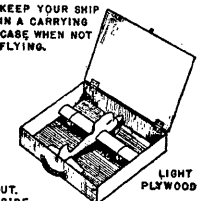
1 & 2 IF BUILDING A SPEED SHIP FROM A KIT OR FROM MAGAZINE PLANS, FOLLOW INSTRUCTIONS TO THE LETTER. WHEN DESIGNING YOUR OWN SHIP, USE ACCEPTED BUILDING METHODS. DON'T SHORT CUT ON STRENGTH & TY.



USE A GOOD FUEL-PROOF FINISH

A GLASSY FINISH WILL ADD UP TO 6 M.P.H. OVER A POOR ONE.

KEEP YOUR SPEED SHIP SCRUPULOUSLY CLEAN BOTH INSIDE & OUT. AFTER EACH FLIGHT REMOVE GRIT & DIRT THAT COLLECTS INSIDE.

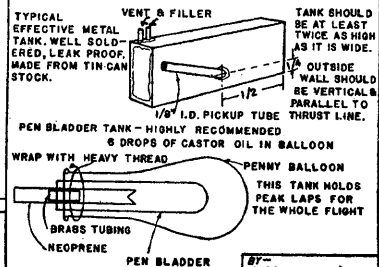
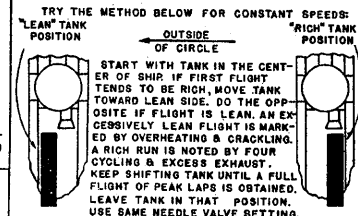


KEEP YOUR SHIP IN A CARRYING CASE WHEN NOT FLYING.

WORK FOR A GOOD FINISH, THEN KEEP IT THAT WAY. USE A HIGH QUALITY PASTE WAX - KEEP IT SHINING

3 YOUR ENGINE, WHEN IT COMES OUT OF THE BOX, IS A PRECISION MECHANISM. KEEP IT THAT WAY. TREAT IT AS YOU WOULD FINE WATCH. BREAK IT IN ACCORDING TO INSTRUCTIONS.

5 A GOOD SPEED TANK IS ONE THAT WILL GIVE YOU PEAK ENGINE PERFORMANCE FOR THE REQUIRED NO. OF LAPS; NO MORE, NO LESS. MANY SPEED FLYERS USE TANKS MUCH TOO LARGE, HOPING TO JUDGE WHEN THE PEAK LAPS WILL OCCUR. BEING GIVEN THE ENGINE UNNECESSARY WEAR, IT IS DIFFICULT TO OBTAIN CONSISTENT PEAK LAPS THIS WAY.



TRY THE METHOD BELOW FOR CONSTANT SPEEDS:

"Lean" TANK POSITION

OUTSIDE OF CIRCLE

"Rich" TANK POSITION

START WITH TANK IN THE CENTER OF SHIP IF FIRST FLIGHT TENDS TO BE RICH, MOVE TANK TOWARD LEAN SIDE. DO THE OPPOSITE IF FLIGHT IS LEAN. AN EXCESSIVELY LEAN FLIGHT IS MARKED BY OVERHEATING & CRACKLING. A RICH RUN IS NOTED BY FOUR CYCLING & EXCESS EXHAUST. KEEP SHIFTING TANK UNTIL A FULL FLIGHT OF PEAK LAPS IS OBTAINED. LEAVE TANK IN THAT POSITION. USE SAME NEEDLE VALVE SETTING.

TANK SHOULD BE AT LEAST TWICE AS HIGH AS IT IS WIDE.

OUTSIDE WALL SHOULD BE VERTICAL & PARALLEL TO THRUST LINE.

1/8" I.D. PICKUP TUBE

6 DROPS OF CASTOR OIL IN BALLOON

WRAP WITH HEAVY THREAD

PENNY BALLOON

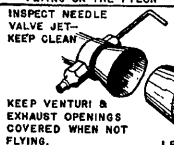
THIS TANK HOLDS PEAK LAPS FOR THE WHOLE FLIGHT

BRASS TUBING

NEOPRENE

PEN BLADDER

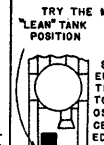
BY HAROLD STEVENSON



INSPECT NEEDLE VALVE SET - KEEP CLEAN



KEEP VENTURI & EXHAUST OPENINGS COVERED WHEN NOT FLYING.



SNUG FIT

HARD BALSA OR PINE PLUGS

NEVER DISASSEMBLE YOUR ENGINE UNLESS REPLACING PARTS. FLUSH IT OUT WITH CLEAN ALCOHOL AFTER THE DAYS FLYING.

4 FUELS ARE GENERALLY THOUGHT TO BE THE "BIG SECRET" OF SPEED FLYING. IN REALITY "SUPER" FUELS MAY HARM YOUR ENGINE. ONCE A GOOD FUEL IS FOUND, STICK WITH IT. CONSTANT CHANGING & EXPERIMENTING LEAVES NO BASIS FOR COMPARING FLIGHT RESULTS IN OTHER FIELDS.



KEEP TIGHTLY CAPPED

STORE YOUR FUEL IN CLEAN GLASS BOTTLES

LABEL YOUR FUELS

USE TOP QUALITY CASTOR OIL IN ALL FUEL FORMULAS (BAKERS AAA)

KEEP FUEL CLEAN

USE GRADUATED GLASS WHEN MIXING FUELS

KEEP TIGHTLY CAPPED

STORE YOUR FUEL IN CLEAN GLASS BOTTLES

LABEL YOUR FUELS

USE TOP QUALITY CASTOR OIL IN ALL FUEL FORMULAS (BAKERS AAA)

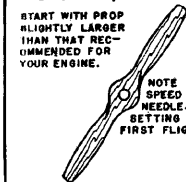
KEEP FUEL CLEAN

USE GRADUATED GLASS WHEN MIXING FUELS

THESE ARE EXCELLENT NAME BRAND RACING FUELS ON THE MARKET BELOW ARE A FEW PROVEN FORMULAS FOR DEPENDABLE USE

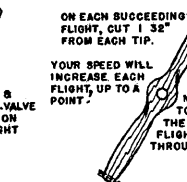
BREAK-IN FUEL	GOOD ALL AROUND FUEL	CONTEST FUEL
2 PARTS METHANOL 1 PART CASTOR OIL	3 PARTS METHANOL 2 PARTS CASTOR OIL	5 PARTS NITROMETHANE 7 PARTS METHANOL 1 PART CASTOR OIL 1 C.C. AMYL AC ETATE PER PINT

IF THERE IS ONLY ONE PROPELLER THAT WILL GIVE PEAK SPEED FOR YOUR PARTICULAR ENGINE-PLANE COMBINATION, NO TWO SPEED JOBS WILL ATTAIN THEIR MAXIMUM SPEEDS WITH EXACTLY THE SAME PROP. BADLY ENOUGH, THE ONLY PRACTICAL WAY TO FIND THE RIGHT PROP. IS THE TRIAL & ERROR METHOD.



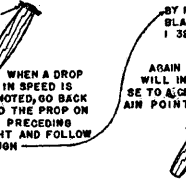
START WITH PROP FLIGHTLY LARGER THAN THAT RECOMMENDED FOR YOUR ENGINE.

NOTE SPEED & NEEDLE VALVE SETTING ON FIRST FLIGHT

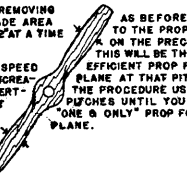


ON EACH SUCCEEDING FLIGHT, CUT 1/32" FROM EACH TIP.

YOUR SPEED WILL INCREASE EACH FLIGHT, UP TO A POINT.



WHEN A DROP IN SPEED IS NOTED, GO BACK TO THE PROP ON THE PRECEDING FLIGHT AND FOLLOW THROUGH



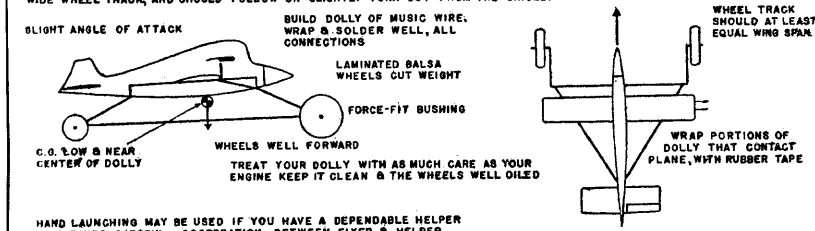
BY REMOVING BLADE AREA 1/32" AT A TIME

AS BEFORE GO BACK TO THE PROP DIMENSION ON THE PRECEDING RUN. THIS WILL BE THE MOST EFFICIENT PROP FOR YOUR PLANE AT THAT PITCH. REPEAT THE PROCEDURE USING OTHER PITCHES UNTIL YOU FIND THE "ONE & ONLY" PROP FOR YOUR PLANE.

AS BEFORE GO BACK TO THE PROP DIMENSION ON THE PRECEDING RUN. THIS WILL BE THE MOST EFFICIENT PROP FOR YOUR PLANE AT THAT PITCH. REPEAT THE PROCEDURE USING OTHER PITCHES UNTIL YOU FIND THE "ONE & ONLY" PROP FOR YOUR PLANE.

WHEN YOU HAVE FOUND THE RIGHT PROP, MAKE UP A GOOD SUPPLY. FINISH YOUR PROPELLERS AS WELL AS YOU FINISH YOUR PLANE. DON'T MAKE BLADES TOO THIN, OR PITCH WILL CHANGE IN FLIGHT. NEVER FLY WITH AN UNBALANCED PROP. ENGINE DAMAGE WILL RESULT.

THE TAKE-OFF DOLLY YOU USE SHOULD HAVE THE FOLLOWING REQUIREMENTS - LIGHT IN WEIGHT, LOW CENTER OF GRAVITY, WIDE WHEEL TRACK, AND SHOULD FOLLOW OR SLIGHTLY TURN OUT FROM THE CIRCLE.



SLIGHT ANGLE OF ATTACK

BUILD DOLLY OF MUSIC WIRE. WRAP & SOLDER WELL, ALL CONNECTIONS

LAMINATED BALSA WHEELS CUT WEIGHT

FORCE-FIT BUSHING

WHEELS WELL FORWARD

TREAT YOUR DOLLY WITH AS MUCH CARE AS YOUR ENGINE. KEEP IT CLEAN & THE WHEELS WELL OILED

WHEEL TRACK SHOULD AT LEAST EQUAL WING SPAN

WRAP PORTIONS OF DOLLY THAT CONTACT PLANE, WITH RUBBER TAPE

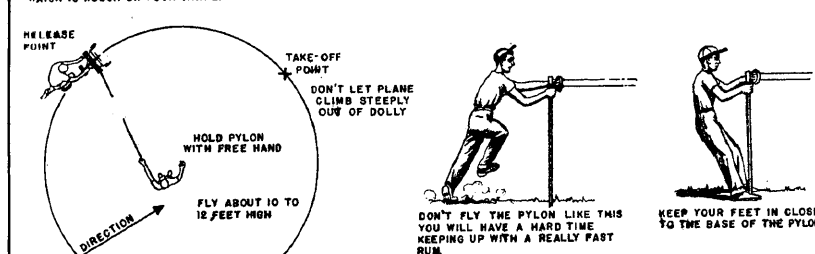
HAND LAUNCHING MAY BE USED IF YOU HAVE A DEPENDABLE HELPER THIS TAKES CAREFUL COOPERATION BETWEEN FLYER & HELPER

DOLLY TAKE-OFFS ARE RECOMMENDED FOR ALL AROUND CONSISTENT PERFORMANCE, ESPECIALLY FOR BEGINNERS

GIVE YOUR DOLLY A COAT OF FUEL PROOF PAINT, TO PREVENT RUSTING & CORROSION

MAKE UP YOUR CONTROL LINES CAREFULLY TO THE EXACT DIMENSIONS CALLED FOR IN A.M.A. RULES. THE LENGTH IS MEASURED FROM THE POINT OF THE HANDLE TO THE CENTER LINE OF THE AIRPLANE. IF YOU ARE OVER THE LENGTH FOR YOUR CLASS, YOU WILL CUT DOWN ON YOUR SPEED. IF YOU ARE UNDER SIZE, YOU WILL NOT BE PERMITTED TO FLY FOLLOW THE DIRECTIONS IN THE A.M.A. RULE BOOK IN MAKING UP CONNECTIONS. KEEP YOUR LINES CLEAN & DRY WHEN THERE IS THE SLIGHTEST DOUBT ABOUT THE SAFETY OF LINES - DON'T FLY.

THE FIRST CONSIDERATION, WHEN GOING OUT FOR A DAYS SPEED FLYING, IS SAFETY. MAKE SURE ALL SPECTATORS ARE WELL BACK FROM THE CIRCLE. ALSO AN EXPERIENCED HELPER IS VALUABLE, IN SEEING THAT PROPER SAFETY MEASURES ARE TAKEN IF POSSIBLE FLY ON A SMOOTH, CLOSE-CROPPED GRASS FIELD. YOU WILL HAVE TO FLY OFF PAVEMENT AT MANY CONTESTS, WHICH IS ROUGH ON YOUR AIRPLANE. SO STICK TO GRASS WHEN PRACTICE FLYING.



RELEASE POINT

TAKE-OFF POINT

DON'T LET PLANE CLIMB STEEPLY OUT OF DOLLY

HOLD PYLON WITH FREE HAND

FLY ABOUT 10 TO 12 FEET HIGH

WIND DIRECTION

NEVER SIGNAL FOR RELEASE UNLESS ENGINE IS RUNNING RIGHT

INSTRUCT YOUR HELPER TO REMOVE THE DOLLY FROM THE FLIGHT PATH AFTER TAKE-OFF. IN LANDING, "FLY" THE MODEL TO THE GROUND. IF SHE GETS DOWN QUICKLY AND SMOOTHLY

AFTER MANY TEST FLIGHTS, YOU WILL KNOW HOW LONG IT TAKES YOUR PLANE TO REACH PEAK SPEED. ANTICIPATE THIS AND GET ON THE PYLON AS SOON AS PEAK SPEED IS REACHED.

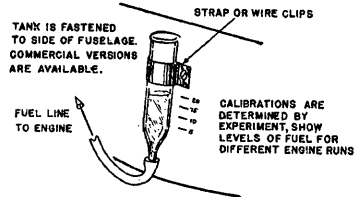
ONCE YOU HAVE IRONED OUT ALL THE "BUGS" IN YOUR SHIP, DON'T FLY EXCEPT IN COMPETITION. IF YOU BREAK THE NATIONAL SPEED RECORD AT YOUR LOCAL FIELD, IT DOESN'T GO IN THE RECORD BOOK. FLY ONLY WHEN IT COUNTS.

ONCE YOU HAVE IRONED OUT ALL THE "BUGS" IN YOUR SHIP, DON'T FLY EXCEPT IN COMPETITION. IF YOU BREAK THE NATIONAL SPEED RECORD AT YOUR LOCAL FIELD, IT DOESN'T GO IN THE RECORD BOOK. FLY ONLY WHEN IT COUNTS.

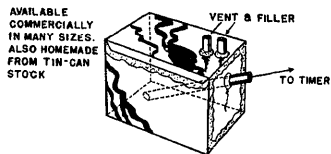
## 28. TANKS & TIMERS

### FREE-FLIGHT

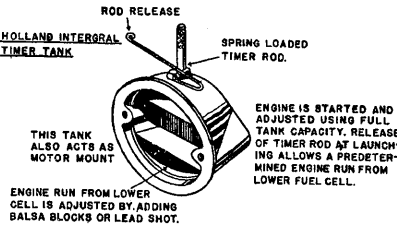
#### EYEDROPPER TANK



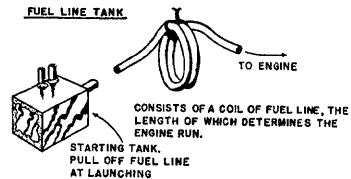
#### STANDARD TYPE F-F TANK



#### HOLLAND INTEGRAL TIMER TANK

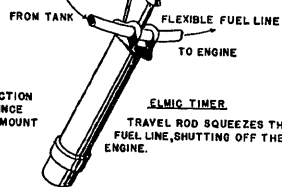
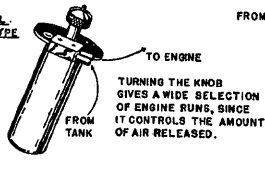


#### FUEL LINE TANK

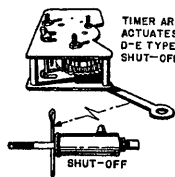


#### TIMERS TO LIMIT ENGINE RUNS

##### TIMER-FUEL SHUT-OFF TYPE



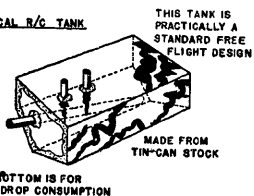
##### CLOCKWORKS TIMER



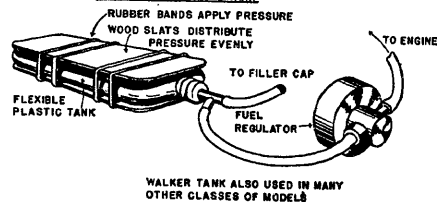
TANKS AND ACCESSORIES MAY BE BOUGHT TO FIT YOUR MODEL

### RADIO-CONTROL

#### TYPICAL R/C TANK



#### JIM WALKER PRESSURE TANK

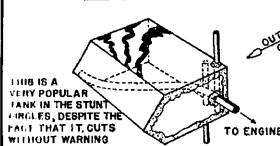


A VARIATION OF THE "CLANK TANK" IS ALSO USED IN R/C (SEE STUNT TANKS)

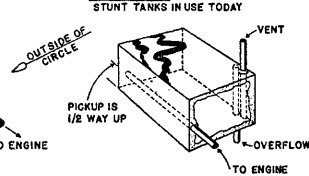
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### CONTROL-LINE STUNT, SPORT & FLYING SCALE

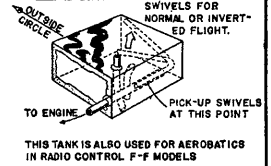
#### TYPICAL "WEDGE" TANK



#### SQUARE TYPE. ONE OF THE BEST STUNT TANKS IN USE TODAY



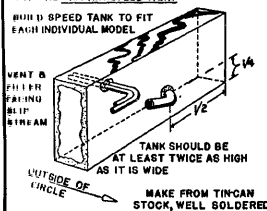
#### "CLANK TANK"



STUNT TANKS MAY BE PURCHASED IN A VARIETY OF SHAPES & SIZES TO FIT YOUR MODELS

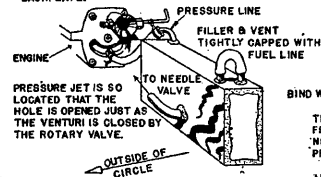
#### WIPED

##### TYPICAL RAM-AIR SPEED TANK

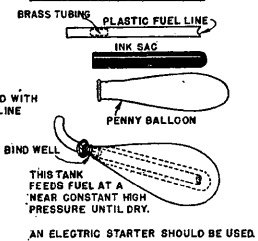


##### TRUE PRESSURE TANK

AIR PRESSURE IS BLED FROM CRANKCASE THROUGH A #60 HOLE DRILLED IN A 4-40 FLAT HEAD MACHINE SCREW TAPPED IN ENGINE BACKPLATE.

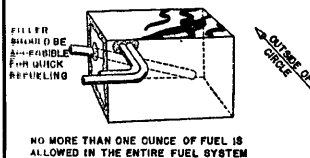


##### PEN-BLADDER PRESSURE TANK

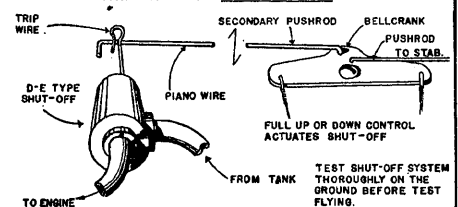


### TEAM RACING

LENGTH X WIDTH X DEPTH MUST BE NO MORE THAN 1.00 CU. IN.

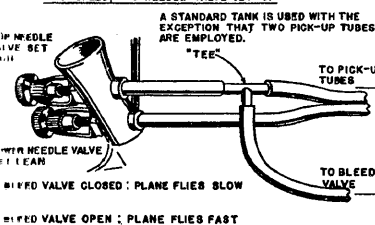


#### TYPICAL TEAM RACER SHUT-OFF SYSTEM



### NAVY CARRIER EVENT

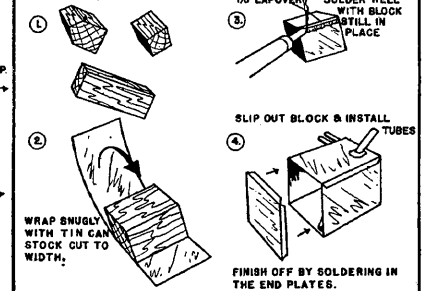
#### TWO-SPEED, TWO NEEDLE-VALVE SET-UP



LOCAL HOBBY-DEALERS CARRY PARTS FOR TWO-SPEED SET-UPS. THIS SYSTEM IS IDEAL FOR SMALL MODELS WHERE SPACE IS AT A PREMIUM.

### TANK CONSTRUCTION

CARVE SHAPE OF TANK FROM WOOD, SLIGHTLY UNDERSIZE



BY HAROLD STEVENSON



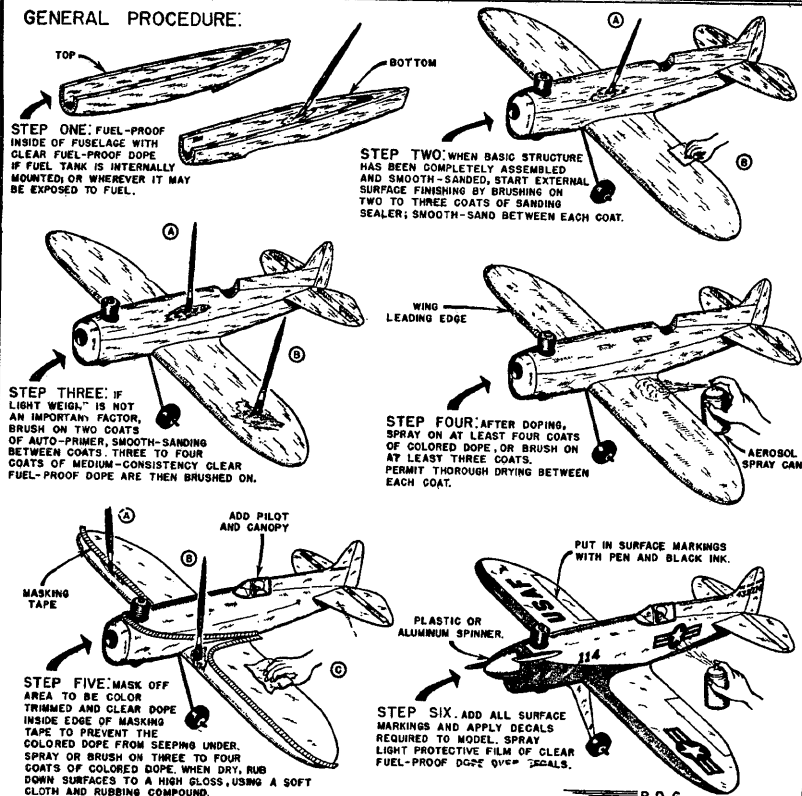
## 29. FINISHING & DECORATING

### WOOD FINISHING:

#### MATERIALS REQUIRED

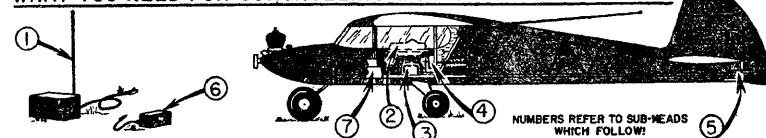


#### GENERAL PROCEDURE:

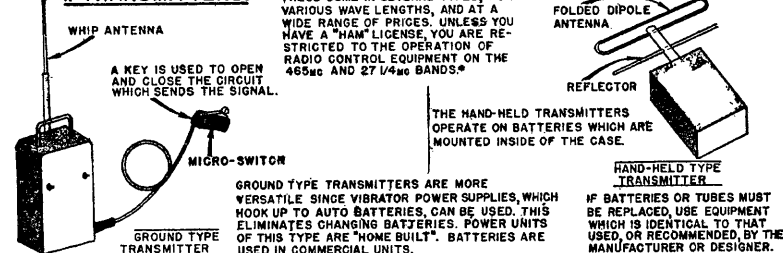


## 30. SINGLE CHANNEL RADIO CONTROL

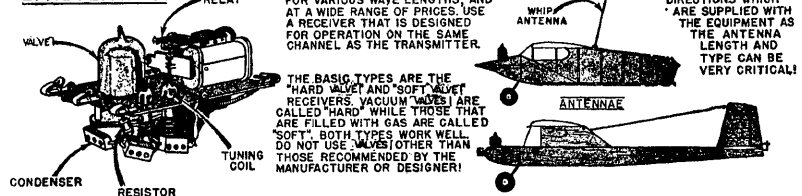
### WHAT YOU NEED FOR CONTROLLING MODEL PLANES BY RADIO:



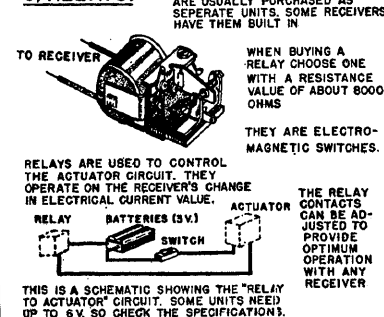
#### 1. TRANSMITTERS:



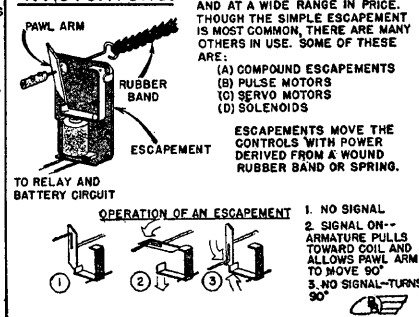
#### 2. RECEIVERS:

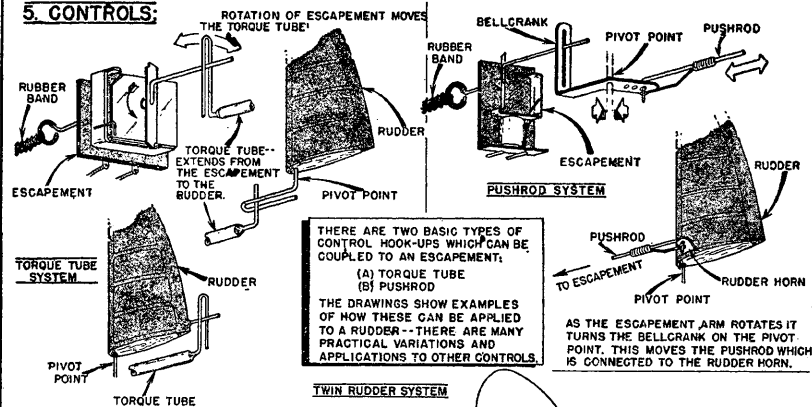
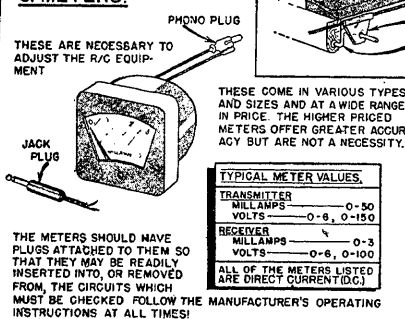
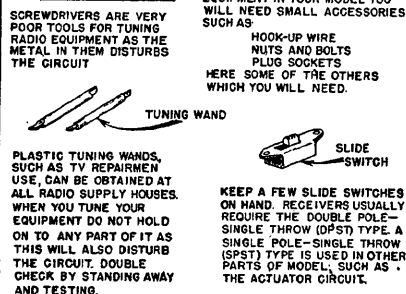
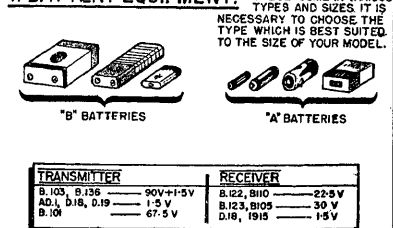


#### 3. RELAYS:

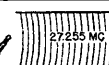


#### 4. ACTUATORS:



**5. CONTROLS:****6. METERS:****ACCESSORIES:****7. BATTERY EQUIPMENT:****31. MULTI-CHANNEL RADIO CONTROL****THE DIFFERENCE BETWEEN SINGLE- AND MULTI-CHANNEL:****SINGLE-CHANNEL:**

THE TRANSMITTER SENDS A 27.255 MC CARRIER WAVE WHICH IS RECEIVED AT THE PLANE. THIS SIGNAL IS CONVERTED INTO A CURRENT WHICH OPERATES A SENSITIVE RELAY. THE RELAY CLOSING AND OPENS A PAIR OF ELECTRICAL CONTACTS THAT OPERATE THE CONTROL CIRCUIT. (SEE 1M DATA SHEET NO. 11 FOR FULL DETAILS)



27.255 MEGACYCLE CARRIER-WAVE IS AT RADIO FREQUENCY AND CANNOT BE HEARD BY THE HUMAN EAR.

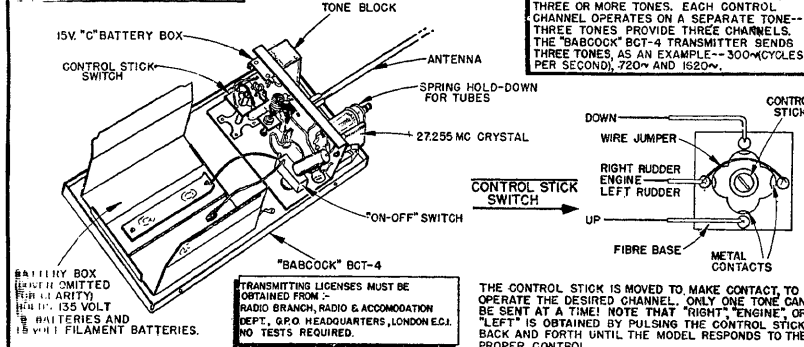
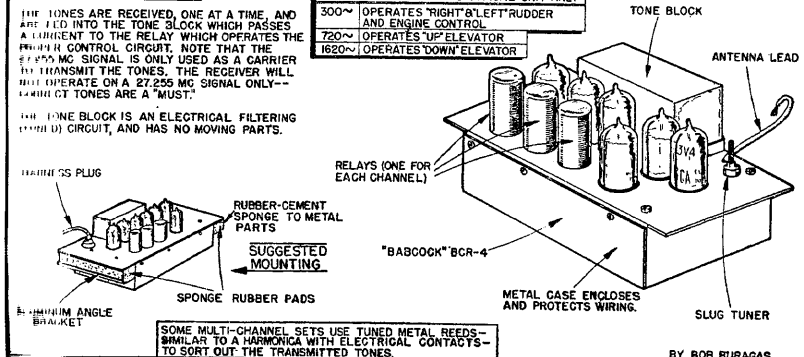
\*OTHER EQUIPMENT IS AVAILABLE FOR COMPLETE MEGACYCLE OPERATION.

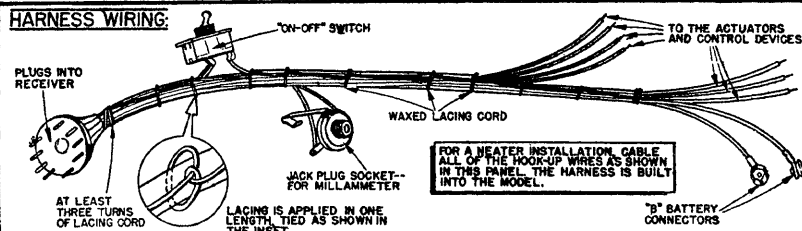
**MULTI-CHANNEL: (TONE-CONTROL)**

A TYPICAL THREE-CHANNEL HAND-HELD TRANSMITTER.

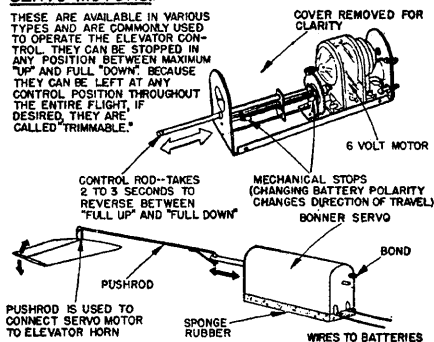


- (1) A 27.255 MC CARRIER-WAVE--SAME AS SINGLE-CHANNEL.
- (2) AN AUDIBLE SOUND OR TONE. THE RECEIVER SORTS OUT THE TONE AND DISCARDS THE 27.255 MC.

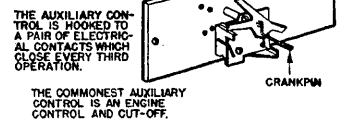
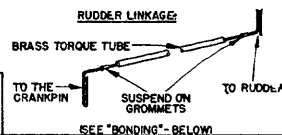
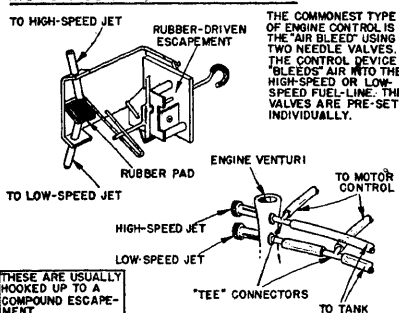
**TYPICAL TRANSMITTER:****TYPICAL RECEIVER:**

**HARNESS WIRING:****SERVO MOTORS:**

THESE ARE AVAILABLE IN VARIOUS TYPES AND ARE COMMONLY USED TO OPERATE THE ELEVATOR CONTROL. THEY CAN BE STOPPED IN ANY POSITION BETWEEN MAXIMUM "UP" AND FULL "DOWN". BECAUSE THEY CAN BE LEFT AT ANY CONTROL POSITION THROUGHOUT THE ENTIRE FLIGHT, IF DESIRED, THEY ARE CALLED "TRIMMABLE."

**COMPOUND ESCAPEMENT:**

THIS IS A RUBBER-DRIVEN UNIT WHICH SUPPLIES RIGHT AND LEFT RUDDER CONTROL PLUS ONE AUXILIARY CONTROL. THE RUDDER IS ACTIVATED BY THE CRANKPIN.

**RUDDER LINKAGE:****MOTOR-CONTROL ESCAPEMENT:****BONDING:**

OPTIMUM PERFORMANCE RESULTS WHEN ALL OF THE METAL PARTS IN THE MODEL ARE BONDED TOGETHER ON A COMMON GROUND, EXCEPT ANTENNA!



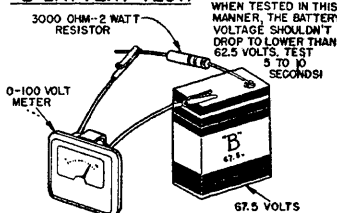
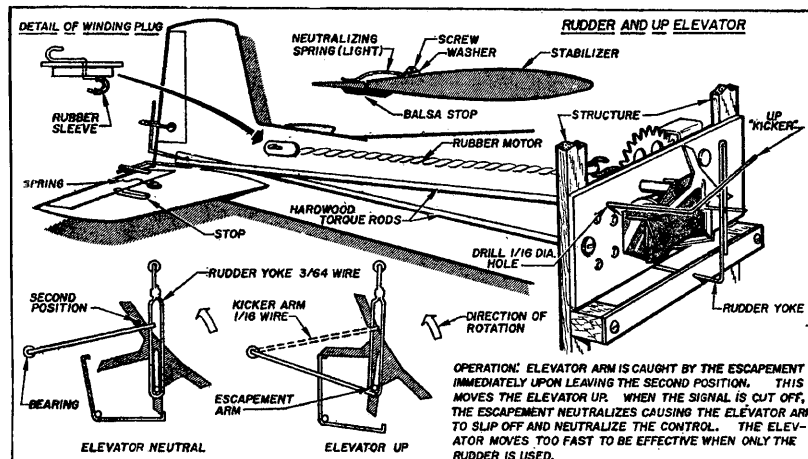
WIRE THE ENGINE, LANDING GEAR STRUTS, ACTUATOR CASES AND FRAMES, PUSHRODS, CONTROL HINGES, RECEIVER BASE, ETC. TO BONDING STRIP

**ACCESSORY EQUIPMENT:**

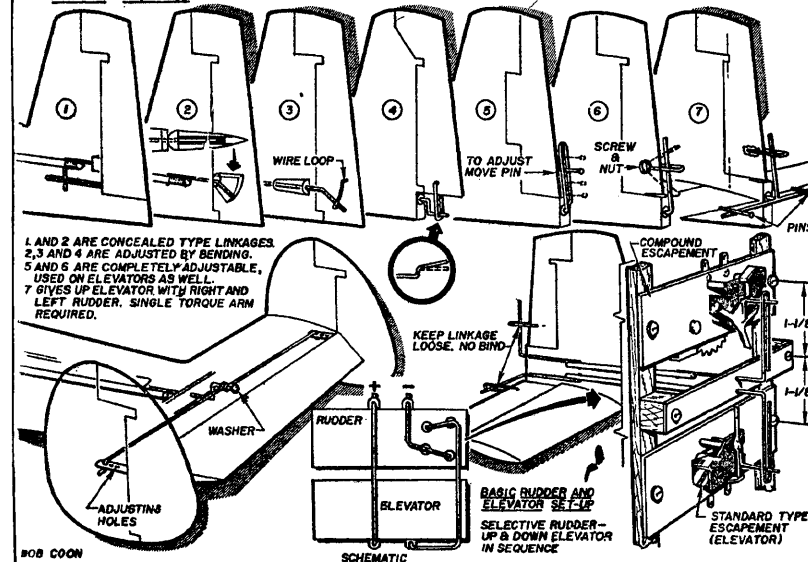
AS WITH SINGLE-CHANNEL R/C EQUIPMENT, IT IS NECESSARY TO HAVE AN ASSORTMENT OF MILLIAMMETERS AND VOLTMETERS--OR A MULTIMETER.

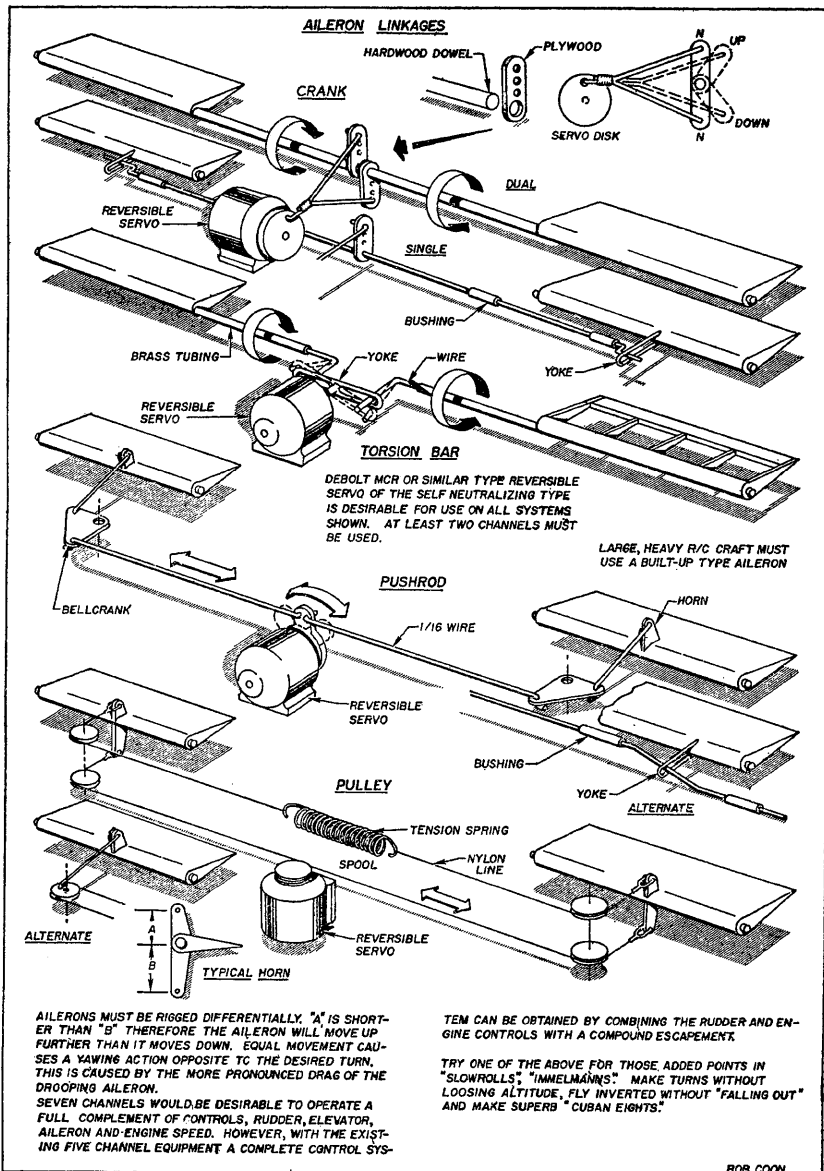
A PLASTIC TUNING WAND IS A MUST FOR BEST RESULTS

HEADPHONES ARE NECESSARY TO TUNE MOST RECEIVERS.

**"B" BATTERY TEST:****32. R/C CONTROL SYSTEMS****RUDDER LINKAGES**

NOTE: DIFFERENT TYPES OF BALANCED SURFACES. NOT TO EXCEED 25% OF TOTAL AREA.



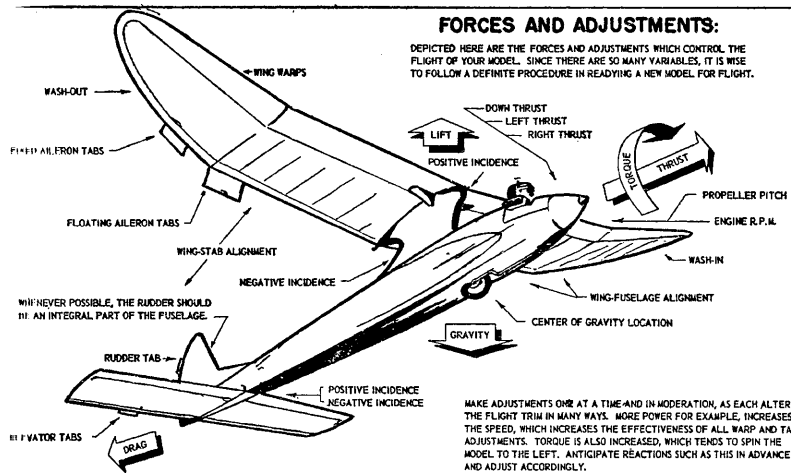


BOB COON

## 33. FREE FLIGHT TRIMMING

### FORCES AND ADJUSTMENTS:

DEPICTED HERE ARE THE FORCES AND ADJUSTMENTS WHICH CONTROL THE FLIGHT OF YOUR MODEL. SINCE THERE ARE SO MANY VARIABLES, IT IS WISE TO FOLLOW A DEFINITE PROCEDURE IN READYING A NEW MODEL FOR FLIGHT.

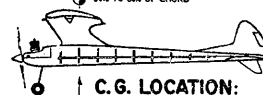


MAKE ADJUSTMENTS ONE AT A TIME AND IN MODERATION, AS EACH ALTERS THE FLIGHT TRIM IN MANY WAYS. MORE POWER FOR EXAMPLE, INCREASES THE SPEED, WHICH INCREASES THE EFFECTIVENESS OF ALL WARP AND TAB ADJUSTMENTS. TORQUE IS ALSO INCREASED, WHICH TENDS TO SPIN THE MODEL TO THE LEFT. ANTICIPATE REACTIONS SUCH AS THIS IN ADVANCE AND ADJUST ACCORDINGLY.

### WASH-OUT:

WHEN THE TRAILING EDGE OF WING TIP UP IS CALLED WASH-OUT. THIS KEEPS THE PLANE FROM STALLING OUT BEFORE THE MAIN PANELS.

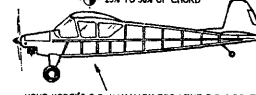
50% TO 80% OF CHORD



### C.G. LOCATION:

THE CENTER OF GRAVITY IS USUALLY LOCATED FURTHER BACK ON THE CHORD ON Pylon TYPES.

25% TO 50% OF CHORD

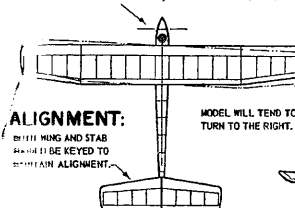


YOUR MODEL'S C.G. MAY VARY FROM THE C.G. LOCATION SHOWN ON THE KIT OR MAGAZINE PLANS, DUE TO SLIGHT DIFFERENCES IN WOOD WEIGHT, ENGINE WEIGHT ETC. THIS IS NOT CRITICAL, AND MAY BE ADJUSTED FOR.

### FLOATING AILERON:

SOME TIMES AN ADJUSTMENT IS REQUIRED FOR CLIMB TRIM ONLY. TRY A FREELY HINGED WEIGHTED CELLULOID TAB. IN CLIMB IT WILL HAVE VERY LITTLE EFFECT.

TEST FLY WITH 0° SIDE THRUST UNLESS OTHERWISE ADVISED.



### ALIGNMENT:

THE WING AND STAB SHOULD BE KEPT TO THE SAME ALIGNMENT.

MODEL WILL TEND TO TURN TO THE RIGHT.

THE WING AND STAB MAY PURPOSELY BE MISALIGNED FOR CORRECTING ADJUSTMENTS. THIS IS NOT FOR BEGINNERS.

### WASH-IN:

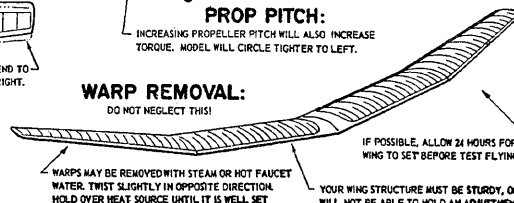
WARPING THE TRAILING EDGE OF WING TIP DOWN IS CALLED WASH-IN. WHEN MODEL IS IN TIGHT CIRCLE, THIS HELPS KEEP INSIDE WING UP.

### PROP PITCH:

INCREASING PROPELLER PITCH WILL ALSO INCREASE TORQUE. MODEL WILL CIRCLE TIGHTER TO LEFT.

### WARP REMOVAL:

DO NOT NEGLECT THIS!



WARPS MAY BE REMOVED WITH STEAM OR HOT FAUCET WATER. TWIST SLIGHTLY IN OPPOSITE DIRECTION. HOLD OVER HEAT SOURCE UNTIL IT IS WELL SET.

IF POSSIBLE, ALLOW 24 HOURS FOR WING TO SET BEFORE TEST FLYING. YOUR WING STRUCTURE MUST BE STURDY, OR IT WILL NOT BE ABLE TO HOLD AN ADJUSTMENT.



### (3) SPIN

### (4) RECOVERY

